



Ka-band InSAR Airborne demonstrator



Final Review

Kick-off meeting

ESA-ESTEC, 11 July 2022



Outline

- Meeting Introduction
- Project Overview
- The KaSAR instrument
- The Aircraft carrier and pod
- The KaSAR processor
- Functional verification campaign
- Requirements review
- Lessons learned and way forward
- Conclusion





adar Solutions

The KaSAR Project

Task 1 - Requirements Review and Consolidation

- Critically review and consolidate/complement requirements
- → Requirements Review (RR)

Task 2 – Preliminary Instrument Design and Performance

- Identify suitable carrier aircraft(s) and define interfaces
- · Define instrument architecture including options
- Define preliminary instrument design
- Develop performance model and derive preliminary instrument performance
- Define processor architecture and algorithms for XTI and ATI image processing.
- Elaborate preliminary on-ground and in-flight verification plan
- → Preliminary Design Review (PDR)

Task 3 - Detailed Instrument Design

- Elaborate detailed instrument design and interface design
- Define procurement approach
- · Create on-ground instrument test plan and procedures
- Create verification test campaign plan and procedures
- Elaborate detailed processor design.
- Update performance model and predicted performance
- → Critical Design Review (CDR)

Task 4 – Instrument Manufacture, Integration and Tests

- Instrument manufacturing, assembly, integration and test
- Interface manufacturing and assembly
- Preparation of the in-flight verification campaign
- Processor implementation
- End-to-end test on ground outside aircraft
- \rightarrow Instrument and Processor (on-ground) Test Review Board (TRB)





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DOCUMENT

Appendix 1 to ESA ITT AO/1-8362/15/NL/FE

Statement of Work

Development of a Ka-band InSAR Airborne Instrument Demonstrator

Task 5 – Verification Flight Campaign and System Verification

- Airworthiness certification
- Instrument integration/accommodation on an aircraft
- Verification Flight Campaign
- System Performance Verification
- Final evaluation of design and test activity and recommendations
- → Final Review (FR)



Goal: Development of a compact, transportable, single-pass, Ka-band InSAR Airborne Instrument Demonstrator (KaSAR).

The KaSAR system is composed of:

- Airborne segment: KaSAR Instrument with control software
- Ground segment: End-to-End Processor from raw data to interferograms

Main development topics of the project:

- 1-Instrument
- 2- Processor
- 3- System verification
- 4- Airworthiness certification



KaSAR is:

- multi-channel (8x) airborne SAR system at Ka-band
- Single-pass Along- and Cross-Track Interferometric SAR
- with Polarimetric SAR capability
- with adjustable and multiple baselines
- Includes the End-to-End Processor from Raw data to Interferograms
- Verified on ground and in-flight
- With airworthiness certification
- Can be used for further airborne SAR data collection experiments



The KaSAR airborne segment







Logical block diagram of the instrument



Mechanical block diagram of the instrument



The KaSAR Instrument

Property	Value
Operating frequency	35.75 GHz
Modulation	FMCW
Bandwidth	500 MHz (adjustable)
Transmit power	10W
Number Tx channels	2 (toggled)
Number Rx channels	8 (simultaneous)
Data storage	8 TB
Antenna size (az x el)	0.1m x 0.05m
Antenna gain	22.3 dBi
Beamwidth (az x el)	4.5° x 35°
Cross-polar	< -27dB
Sidelobe level	< -15dB
Polarization	Dual linear (H & V)

Property	Value
Height of operation	500m < h < 3000m
Incidence angles	20° - 50°
Operational modes	Single-pol V, Single-pol H, full-pol V&H, ping-pong, dual-view L/R
Adjustable Baselines (RX antenna distance)	0.1m < ATB < 0.5m 0.1m < XTB < 0.4m
NESZ	<-20dB
Swath width	> 1km (resolution dependent)
Ground resolution	< 1m x 1m (SLC)
Inter-channel phase accuracy	< 1° RMS (after I-CAL and E-CAL corrections)
Vertical accuracy (from the data)	~10cm flat areas (high coherence) ~50cm forested areas
Spaceborne reference scenario	400-800 km orbit height, 7 km/s speed, 16 km swath (NESZ< -16dB)



The KaSAR Instrument

Technology Safety Data Sheet KaSAR Synthetic Aperture Radar

Section 1: Technology Identity						
Technology Name(s):	Emergency Contact:				
Synthetic Aperture Rada	r	Christian Trampuz Huygensstaat 44 2201 DK Noordwijk The Netherlands shristian trampuz@metasensing.com +31717615092				
Manufacturer's Nar	ne and Address:	Information Contact:				
MetaSensing BV Huygensstraat 44 2201DK Noordwijk The Netherlands		E-mail: info@metasensing.com Phone: +31717515000				
Date Prepared:	December 2021	Date Revised:	Not yet revised.			

Section 2: Technology Pictures





Figure 1: Radar and 3-axis stabilizer under test.

KaSAR - 2021-12 Rev. B











Command & Control Software

TROL DATA PREV	VIEW						
SYS	TEM	ADC	AWG	RF1	IMU	GIMBAL	
RE4	NDY .	READY	READY	DISCONNECT	READY	READY	IP: 127.0.0.1 Disconnect
IPERATURES					HUMIDITY		PRESET
T_TX_1	T_TX_2	T_RX_A	T_RX_B	T_RX_C	H_TX_1	H_TX_2	Folder: D:/KaSAR/oreset
0.0 ℃	0.0 ℃	0.0 °C	0.0 °C	0.0 °C	0.0 %	0.0 %	
AWG	ADC	RF1	RF2		SENSOR3		Preset: monopoi_v
42.0 °C	39.0 °C	0.0 °C	0.0 ℃		0.0 %		LOAD PRESET
I-GP5				RF			TRACKS
YAW	ROLL	РІТСН	GROUND SPEED	MODE	POWER TX-V	POWER TX-H	Tracks file: D:/
334.457°	-0.007°	0.120°	0.0 Km/h	STAND_BY	0.0 dBm	0.0 dBm	
LAT	LON	ALT	SATELLITES				
52.213127	4.429377	12.8 m	15 (16)				PRF Desired Bandwidth RF
		DICH			686		4000 Hz V 500 MHz V TX Enab
AWG	ADC	DATA RATE	AVAILABLE SPACE	REMAINING TIME	UTC	TIME	PRF Actual Resolution Max Range
LOCKED	LOCKED	396.6 MB/s	4838 GB (63%)	201 min	23 Nov 202	0 - 13:12:45	4000 H2 MHZ M
power_dB=20 reduction_offs 20-11-23 14:1).000000 if_atten et from lsb=0	uation_dB=0.0000	100 tx_enabled=1 b	itreduction_enabl	ed=0	^	
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Configuration and operation GUI with Pilot Assistant





Command & Control Software

In-flight acquired data monitoring







The aircraft carrier with pod











METASENSING Radar Solutions









Radome structure A-sandwich

Typical performance





Scaled to Ku-band (13.5 GHz) and Ka-band (35.75 GHz)





The KaSAR Processor



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From Raw data to SAR Interferograms





The KaSAR Processor

Configuration file

GUI

filled_kasar_cfg.json - Notepad	KaSAR Focuser			-	- 🗆 🗙
File Edit Format View Help					
<pre>{ "description": "Input configuration file", "version": "1.0.0", "datatake_path": "Y:/ATI_ADD_INKUX/new_data/Ku-band", "sensors_log_path": "Y:/ATI_ADD_INKUX/new_data/Ku-band", "output_path": "Y:/ATI_ADD_INKUX/new_data/Ku-band", "preprocess": { "rc_window": "HAMMING", "ical_averaging_sec": 100.0 }, " </pre>	Preprocessing Range Compression Window Internal calibration average [sec]	Kemming ~	DEM Mode Value/path	SRTM Browse V:\SRTM	~
<pre>"focuser": { "look_direction": "left", "grid_geometry": "ground", "antenna_beamwidth_deg": 0.0, "upsampling_factor": 4.0, "grid": { "range_min_m": 476.0, "range_max_m": 1895.0, "range_spacing_m": 1.0, "azimuth_spacing_m": 1.0 }, "az_window": "HAMMING", "data": { "data"; "data";</pre>	Focuser Look direction Left Grid geometry Ground Antenna beamwidth (deg) 0 Upsampling factor 4		Grid Range min [m] [Range max [m] [Range spacing [m] [Azimuth spacing [m] [Azimuth Window [476 1895 1 1 Hamming	· · · · · · · · · · · · · · · · · · ·
"mode": "srtm",	Packaging		Configuration		
<pre>"path": "V:\\SRIM" } , "packaging": { "interferometry": { "looks": { "range_pixels": 5,</pre>	Interferometry range pixels		Y:/ATI_ADD_INKUX/n	ew_data/Ku-band ew_data/Ku-band	Browse
"azimuth_pixels": 5	and reformed y damage pixels				
}	Radiometric res range pixels 5		Y:/ATI_ADD_INKUX/n	ew_data/Ku-band	Browse
}, "radiometric_res": { "looks": { "range nivels": 5	Radiometric res azimuth pixels 5				
"azimuth_pixels": 5 } }	Configure Data saved in kasar_cfg.json	Pad	kage	Run	



The processing of the Functional campaign data this presentation has been done interfacing with the Configuration text file and Running one single call:

kasar_processor ('C:\kasar_processor\gui\kasar_cfg0.json', 'focuser')

```
"description": "Input configuration file",
"version": "1.0.0",
"datatake path": "V:\\KASARFUNCCAMP PROJECT\\20220504T134817",
"post processed nav": "V:\\KASARFUNCCAMP PROJECT\\postprocnav\\flight2\\flight2.txt",
"output path": "V:\\KASARFUNCCAMP PROJECT\\",
"preprocess": {
    "rc window": "HAMMING",
    "ical averaging sec": 1.0
},
"focuser": {
    "look direction": "left",
    "grid geometry": "ground",
    "antenna beamwidth deg": 0.41,
    "upsampling factor": 4.0,
    "grid": {
        "range min m": 500.0,
        "range max m": 2500.0,
        "range spacing m": 1.0,
        "azimuth spacing m": 1.0
    },
    "az window": "HAMMING",
    "dem": {
        "mode": "mean",
        "value": 0.0
},
"packaging": {
    "interferometry": {
        "looks": {
            "range_pixels": 0,
            "azimuth pixels": 0
    },
    "radiometric res": {
        "looks": {
            "range pixels": 1,
            "azimuth pixels": 1
    }
3
```



The Processor allows interfacing to every module independently.

The processor works running 3 independent procedures:

- 1. kasar_preprocessor.m (1 call in Matlab)
- 2. Pre_nav, nav, az_comp (procedure made of 3 calls..3 c++ executables)
- 3. MAIN_KaSAR_packager.m (1 call in Matlab)



The KaSAR Processor

The Pre-Processor

The Pre-Processor does the data transcription and range compression

1. Set the schema of the KASAR pre-processor

input_schema_fname = '/home/metasensing/repos/kasar_processor\gui\schemas\v4.schema.json';





The Focuser

- 1) The Focuser begins by ingesting the post-processed navigation data and interpolating them into the time grid of the raw data, so the navigation and range compressed data are synchronized.
- 2) The actual focusing starts afterwards by running the code that implements the GBP algorithm





The Packager

- The Packager outputs the interferometric SAR data and corresponding metadata in one single Netcdf file.
- The Netcdf are saved in pairs of interferometric data (master and slave)
- All available combination of channels are outputted.
- Additionally, visualization of intensity, radiometric resolution and interferometric phase and coherence are made available in .jpeg format and .fig



The KaSAR Processor Outputs:

Data from the Functional Campaign showing the system performance



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Verification Functional Campaign Description



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Functional Campaign flown on:

- 02 May 2022
- 04 May 2022









The functional campaign has Ground Control Points used to quantitatively verify the performance of the processor in terms of resolution, radiometric and phase accuracy.







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Functional Campaign: Position of the antennas







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Look direction

Functional Campaign: Position of the antennas and modes:





- Number of acquisitions: 40

2 May 2022: Single-Pol-V: 3 Single-Pol-H: 3 4 May 2022: Single-Pol-H: 7 Ping-pong H: 6 Single-Pol-V: 4 Ping-pong V: 4 Full-Pol: 13

Example of log file

Acquisition name	Operational mode	BW [MHz]	PRF [kHz]	Num channels	Gimbal	Description
Flight 1 (02 may)						V-pol antennas connected
20220502T155925	Single-Pol V	500	4.0	4	Compensation on	Test acquisition
20220502T160050	Single-Pol V	500	4.0	4	Compensation on	Test acquisition
20220502T160132	Single-Pol V	500	4.0	4	Compensation on	Test acquisition
Flight 2 (02 may)						H-pol antennas connected
20220502T181930	Single-Pol H	500	4.0	4	Compensation on	Test acquisition
20220502T182031	Single-Pol H	500	4.0	4	Compensation on	Test acquisition



With the functional campaign data, we aim to fully test the KASAR Processor in all operation modes (V, H, full-pol, ATI, XTI) and demonstrate that the functionalities are implemented.

Therefore, the following tracks has been selected:

20220502T172210 (Single-Pol H) 20220504T134817 (ping-pong V) 20220504T141735 (Single-Pol V) 20220504T163000 (full-pol)

Note that the name of the processed data might differ from raw data because the name of the processed file comes form the gps time, which is in fact the time of start of the acquisition.



Before Proceeding – Check Raw data



20220504T163000, ch 211 624 B -1650.165 1289.1914 R 100.7303 1906.68 60 500 55 **Cross-pol** Slant Range - [m] 1200 1200 50 45 40 35 30 2000 25 -3000 -2000 -1000 0 1000 2000 3000 Doppler - [Hz]

KaSAR Processor Outputs









KaSAR Processor Outputs

Before Proceeding – Check Nav



KaSAR Processor Output & Analysis

Radiometry

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Single-Pol (HH)

Ping-Pong (VV)

KaSAR Processor Outputs

Single-Pol (VV)

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KaSAR Processor Outputs

Full-Pol

ASENSING

Radar Solutions

M

ΗH

Full-Pol

RGB Composite (Red= HH, VV=Blue, HV=Green) (3x3 m reso, 1km swath) (20220504T163000 (channels 210 (HH), 111(VV), 110(VH))

Corner Response only in HH and VV (Red + Blue = Purple)

25 dB between Co-pol /Cross-pol ratio

KaSAR Output Analysis

Corner Reflector IRF

SING

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Corner Reflector IRF

KaSAR Output Analysis

NESZ

NESZ < -20 dB

(approx. - 25 dB)

Based on low Backscatter areas In the sigma-0 Calibrated image

KaSAR Processor Output & Analysis XTI

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XTI Single-Pol (HH)

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XTI Single-Pol (HH)

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XTI Single-Pol (VV)

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XTI Single-Pol (VV)

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1km

XTI Ping-Pong (VV)

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XTI Ping-Pong (VV)

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KaSAR Output Analysis

KaSAR Output Analysis

SAR_CPLX_20220502172210_11_pres_16_fdc_-310.sar.pow.sm2x4.db.dec.geo

COH_20220502172210_12_pres_16_fdc_-310_X_20220502172210_11.interf.geo

Legend

PHA_20220502172210_12_pres_16_fdc_-310_X_20220502172210_11.interf.geo

Flat phase

Very high coherence (~1)

KaSAR Processor Output & Analysis ATI

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ATI Single-Pol (VV)

Radar Solutions

M

ATI Single-Pol (VV)

Requirements review - General

ITEM ID in SOW	TEST ID In D11	PURPOSE	YES/NO	MILESTONE
GEN-010	T01-GEN-POS	The airborne demonstrator shall be representative of the spaceborne reference scenario.	YES	TRB-1/TRB-2
GEN-020	T02-GEN-POS	The minimum covered incidence angle range shall be 20- 50 deg.	YES	FR
GEN-030	T03-GEN-POS	The minimum swath width shall be 1 km in stripmap mode.	YES	FR
GEN-040	T04-GEN-LAB	A least 1 suitable aircraft to carry the instrument shall be identified.	YES	TRB-2
GEN-050	T05-GEN-AIR	The instrument shall allow fast and simple (de-) installation on the aircraft.	YES	FR
GEN-060	T06-GEN-AIR	The Instrument shall be designed with suitable well documented control interfaces	YES	FR
GEN-070	T07-GEN-POS	The Instrument shall be suitable for usage in polar and tropical climate areas.	YES	FR
GEN-080	T08-GEN-LAB	The instrument shall be built in a way to allow for easy accommodation and/or implementation of additional components/functionality.	YES	TRB-2
GEN-090	T09-GEN-POS	The instrument shall allow to emulate the performance of XTI spaceborne case	YES	FR

Requirements review - Instrument

ITEM ID in SOW	TEST ID	PURPOSE	YES/NO	Verification
INS-010	T010-INS-LAB	The instrument center frequency shall be Ka-band at 35.75 GHz.	Yes	TRB-1 / TRB-2
INS-020	T020-INS-POS	The instrument shall be designed to allow SLC 1m x 1m (el. x az.) on ground resolution.	Yes	FR
INS-025	T020-INS-LAB	The instrument shall be designed to allow SLC 1m x 1m (el. x az.) on ground resolution (bandwidth, antenna size).	Yes	TRB-1/TRB-2
INS-030	T030-INS-POS	The instrument shall be designed for NESZ of less than -20dB for VV/HH polarization in clear sky conditions.	Yes	FR
INS-040	T040-INS-POS	The instrument shall be designed for a DTAR of less than -25dB for VV/HH polarization in clear sky conditions.	Yes	FR
INS-045	T045-INS-POS	The instrument shall be designed for a radiometric error (*) of 0.5 dB (1-sigma) and 1 dB (3-sigma). Degradations up to 0.8 dB (1-sigma) and 1.3 dB (3-sigma) will be accepted.	Yes	FR
INS-050	T050-INS-LAB	The instrument shall be designed for a cross polar isolation of -25dB	Yes	TRB-1
INS-055	T055-INS-INF	The instrument shall be designed for a cross polar isolation of -25dB	Yes	FR
INS-060	T060-INS-LAB	The instrument shall have 4 Rx channels	Yes	TRB-1
INS-070	T070-INS-LAB	The instrument shall be extended to have a total of 8 Rx channels	Yes	TRB-1
INS-080	T080-INS-LAB	The antenna system shall support simultaneous H and V	Yes	TRB-1
INS-090	T090-INS-LAB	The instrument shall be easily extendable to HH, HV, and VV, VH operation	Yes	TRB-1
INS-100	T100-INS-LAB	Observation concept: Single-Pass Interferometric SAR	Yes	PDR/CDR
INS-110	T110-INS-LAB	XTI: multiple baselines in cross-track	Yes	TRB-2

Requirements review - Instrument

ITEM ID in SOW	TEST ID	PURPOSE	YES/NO	Verification
INS-120	T120-INS-LAB	ATI: multiple baselines in along-track	Yes	TRB-2
INS-130	T130-INS-LAB	Hybrid configuration between XTI and ATI	Yes	TRB-2
INS-140	T140-INS-LAB	Extended functionality: +/- 45 deg squint, enhanced polarimetry	Yes	TRB-2
INS-150	T150-INS-INF	Attitude and position knowledge and control	Yes	FR
INS-155	T155-INS-LAB	Attitude and position knowledge and control verification on ground	Yes	TRB-2
INS-160	T160-INS-POS	Suitable internal and external calibration (inter-channel phase accuracy of 1 deg RMS)	Yes	FR
INS-165	T165-INS-LAB	Suitable internal calibration (inter-channel phase accuracy of 1 deg RMS)	Yes	TRB-2
INS-166	T166-INS-LAB	Suitable temperature and humidity monitoring instruments shall be installed in the pod near the antennas	Yes	TRB-2
INS-170	T170-INS-LAB	On-board data storage for at least 2 hours	Yes	TRB-1 / TRB-2
INS-180	T180-INS-POS	Ancillary data (navigation, calibration data) should be available and recorded	Yes	FR
INS-185	T185-INS-LAB	Ancillary data (navigation, calibration data, temperature and humidity) should be available, recorded and meaningful	Yes	TRB-2
INS-190	T190-INS-INF	In-flight operation monitoring	Yes	FR
INS-195	T195-INS-LAB	Real-time monitoring of correct instrument functioning	Yes	TRB-2
INS-200	T200-INS-INF	Optical Swath monitoring	Yes	FR
INS-210	T210-INS-POS	Adjustable signal bandwidth for low resolution modes	Yes	FR
INS-215	T215-INS-LAB	Adjustable signal bandwidth for low resolution modes	Yes	TRB-1
INS-220	T220-INS-LAB	The fully integrated instrument operates correctly and generates the expected outputs on the ground (radar and ancillary data)	Yes	TRB-2 56

Requirements review - Processor

ITEM ID in SOW	TEST ID	PURPOSE	YES/NO	MILESTONE
PRO-010	T010-PRO-LAB	The processor shall be developed for interferometric phase maps representation and independent of the carrier platform	Yes	TRB-2
PRO-020	T020-PRO-LAB	As input, the processor shall accept the raw data in the format as acquired by the instrument and without the need for any pre-processing Output in netCDF of map of radiometric resolution, interferometric coherence maps, multi-looked interferometric phase maps	Yes	TRB-2
PRO-030	T030-PRO-POS	Non-free licenses shall not be used, unless necessary for improving the performances of the processor	Yes	TRB-2
PRO-040	T040-PRO-POS	The processor shall have a simple Graphical User Interface	Yes	TRB-2
PRO-050	T050-PRO-POS	It shall be possible to set all simulator parameters by means of a user-friendly configuration file.	Yes	TRB-2
PRO-060	T060-PRO-POS	A set of graphic routines supporting all the operations required for displaying and printing the graphical outputs shall be provided.	Yes	TRB-2
PRO-070	T070-PRO-POS	The source code of the processor shall be provided	Yes	TRB-2
PRO-080	T080-PRO-POS	The source code of the tool shall be written in such a manner that it may be understood, modified and debugged by typical end users	Yes	TRB-2
PRO-090	T090-PRO-POS	The source code shall be written in an appropriate way to allow future modifications.	Yes	TRB-2
PRO-100	T100-PRO-POS	The processor shall be easily extendable	Yes	TRB-2
PRO-110	T110-PRO-POS	The processor shall be designed platform independent	Yes	TRB-2

Lessons Learned & way forward

- Complex development project of a state-of-the-art radar system
- Multipath and coupling within the pod
- Handling of multiple channels simultaneously generating large amount of data to record on disk
- Airworthiness certification for the radar and the pod
- Project management to correctly handle all the complex development
- Importance of sub-contractor management
- The KaSAR system is ready to be operated for dedicated airborne SAR data collection campaigns for different studies.

Conclusions

- The objective and all the tasks within the project have been reached and performed.
- The KaSAR system (instrument + processor) has been developed, operated and evaluated
- KaSAR is a multi-channel, airborne, polarimetric, single-pass Along- and Cross-Track Interferometric SAR (InSAR) at Ka-band, with adjustable and multiple baselines.
- The instrument is supported by the End-to-End SAR processor.
- The sensor is hosted on MetaSensing's PA31 aircraft.
- A dedicated pod has been designed, fabricated, integrated on the aircraft, certified and validated in flight in terms of RF performance.
- The functional verification campaign has been flown on 2nd and 4th May 2022 and data have been processed and analyzed.
- Successful results have been obtained showing that the instrument and the processor can be used operationally for further Ka-band data collection campaign experiments.

Conclusions

- Updates and some improvements can be introduced in the processing software.
- The first release of the KaSAR processor is operational and its functionality has been tested with the functional campaign data.
- The outputs of the KaSAR processor shows that all modes of operations of KaSAR instrument can be ingested and processed.
- Further refinement in the accuracy and quality of the KaSAR processor output data will be tackled having the MetaSAR-Pro as source of comparison and analyzing a greater volume of data.
- All data acquired and here presented are delivered as raw data, together with the delivered processor and configuration files, so that all results presented and obtained are reproducible.

Thank you!

