

Final Presentation to ESA

**MIRROR:  
A RE-LOCATABLE  
MANIPULATOR FOR ON-ORBIT  
ASSEMBLY AND SERVICING**

**mirror**  
Multi-arm Installation Robot for Reaching ORUS and Reflectors



 **LEONARDO**

**iiT** ISTITUTO  
ITALIANO DI  
TECNOLOGIA

 **SENER**  
Aeroespacial

**gmV**  
INNOVATING SOLUTIONS

# Introduction

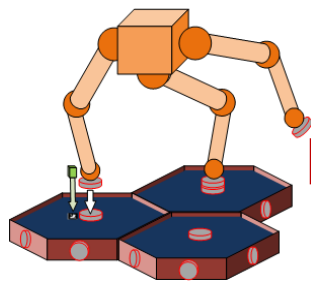


- **Multi-arm Installation Robot for Readying ORUS and Reflectors (MIRROR)**
- ESA contract covering preliminary design (flight system) and development and ground testing of a breadboard demonstrator (TRL4)
- Industrial Consortium:

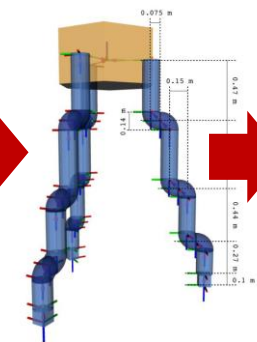


# MIRROR System Development

## Development process



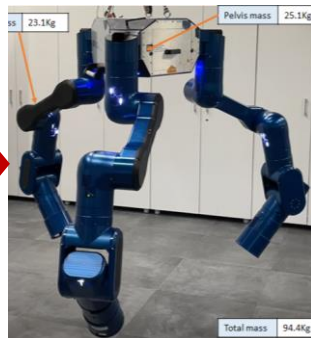
Requirements &  
CONOPS  
**SRR June 2020**



Preliminary Design  
of Flight System  
**PDR Nov 2020**



Detailed Design of  
Breadboard System  
**BDR April 2021**



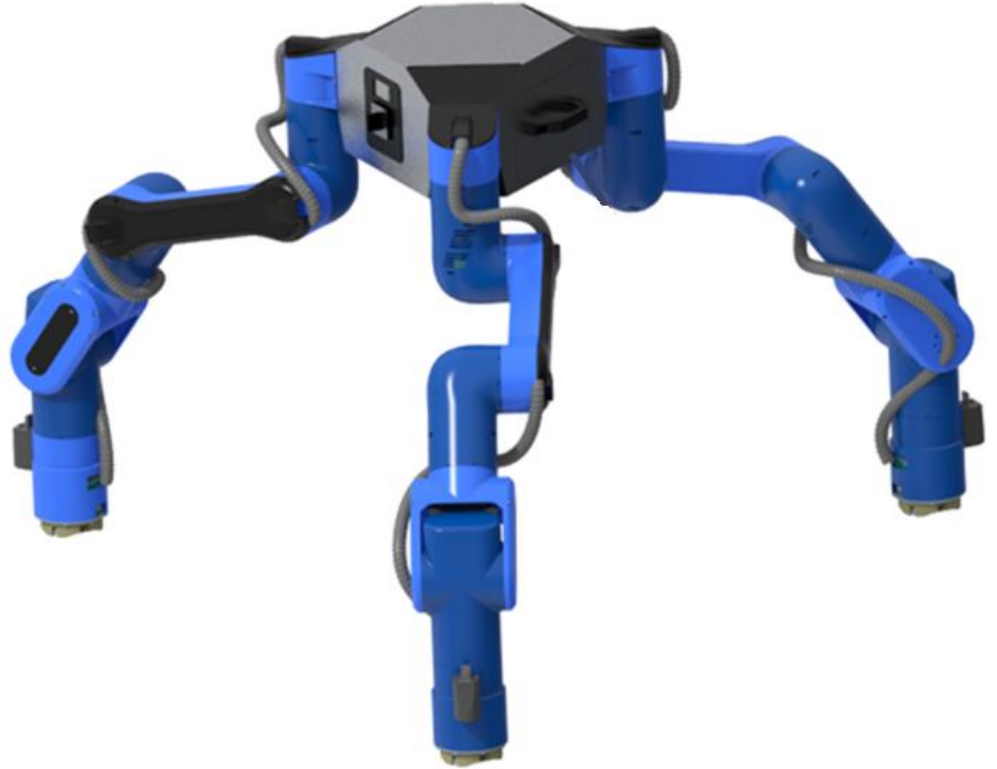
Manufacturing and  
Integration  
**TRR July 2022**



Ground testing of  
Breadboard System  
**FA July 2023**

# Contents

- Mission and Concept of Operation
- Breadboard System Design
- Test Plan
- Testbed Design
- Test Results
- Conclusions

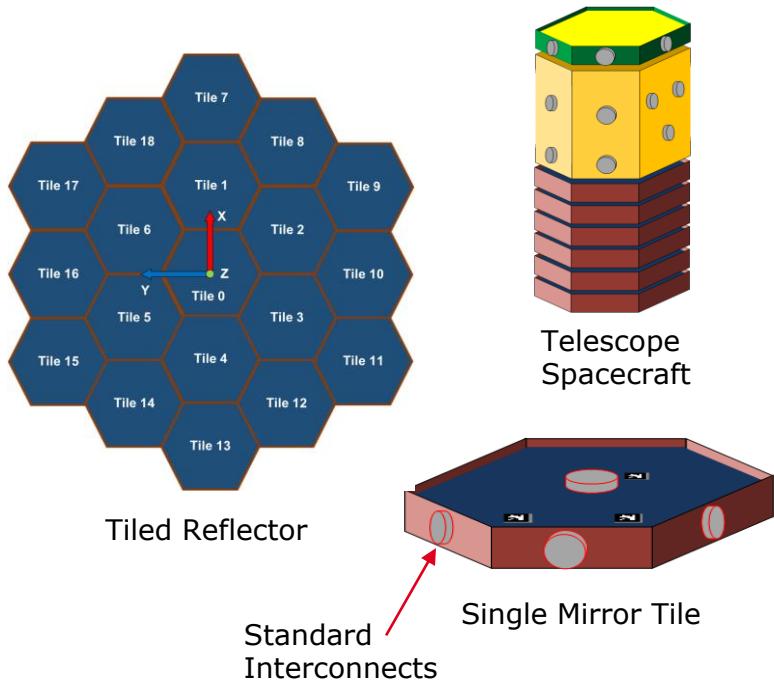


# Mission and Concept of Operation (I)

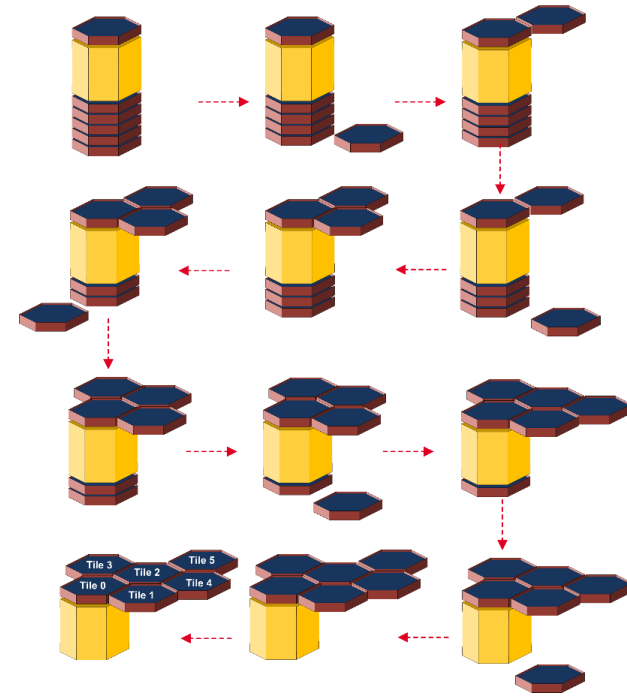


## Scenario

On-orbit Assembly of a Multi-Ring Telescope Reflector



Launch Configuration



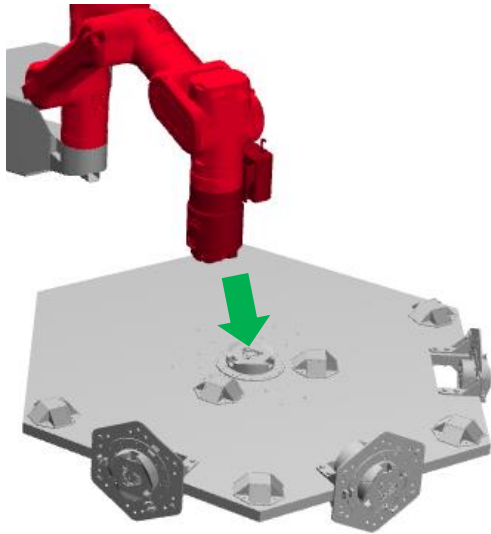
On Orbit Assembly

# Mission and Concept of Operation (II)

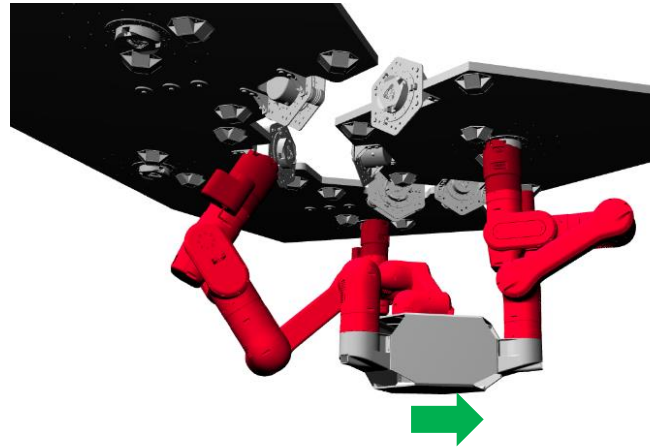


## Concept of Operation

Multi-Arm Relocatable Manipulator equipped with **three arms** capable of three main operations:

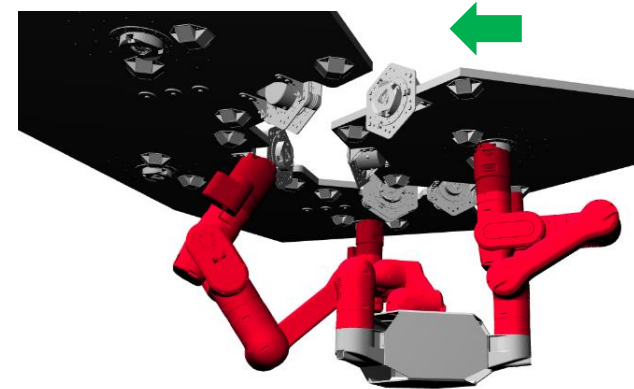


**Tile Manipulation**



**Locomotion & Tile Transport**

Walk with two arms, eventually use the third one to transport a tile



**Tile assembly**

# Contents

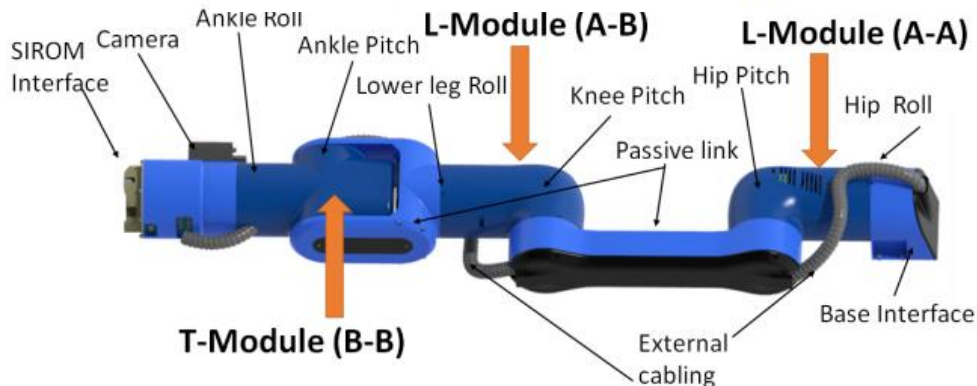
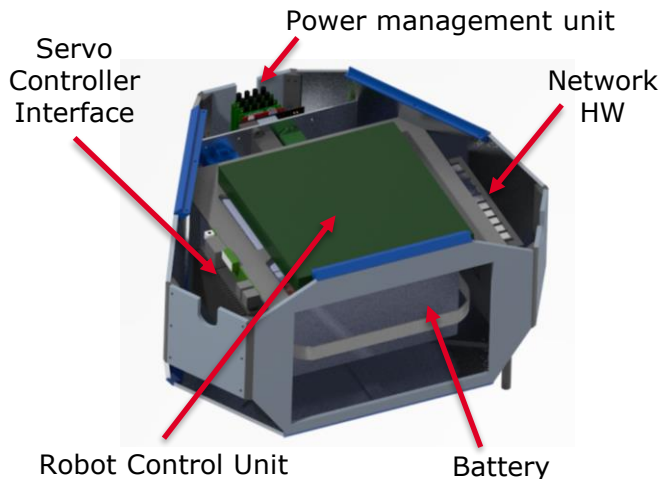
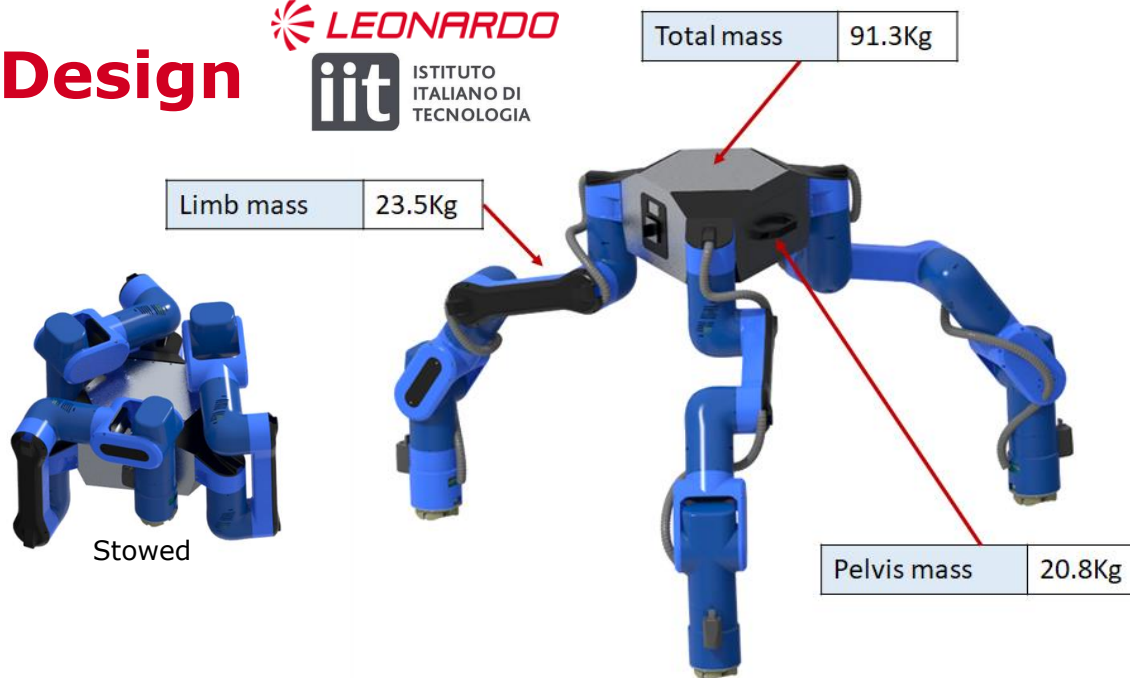
- Mission and Concept of Operation
- Breadboard System Design
- Test Plan
- Testbed Design
- Test Results
- Conclusions



# Breadboard System Design

## Relocatable Manipulator

- Exoskeleton structure including:
  - Pelvis
  - Three identical 6DOF arms
  - End effectors → SIROM

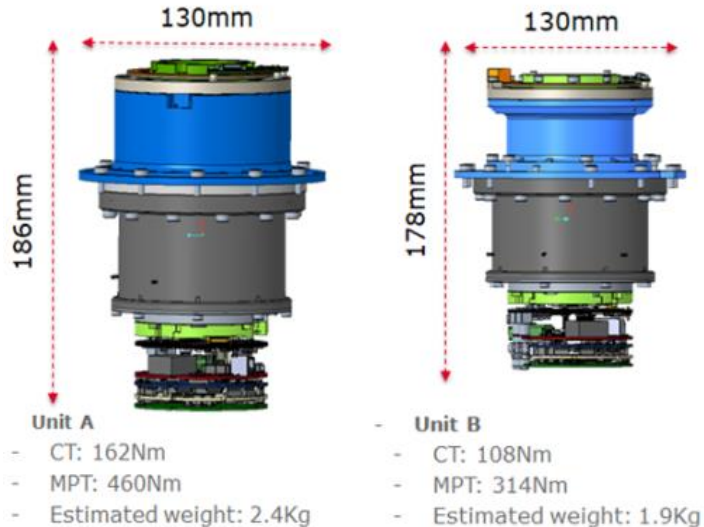
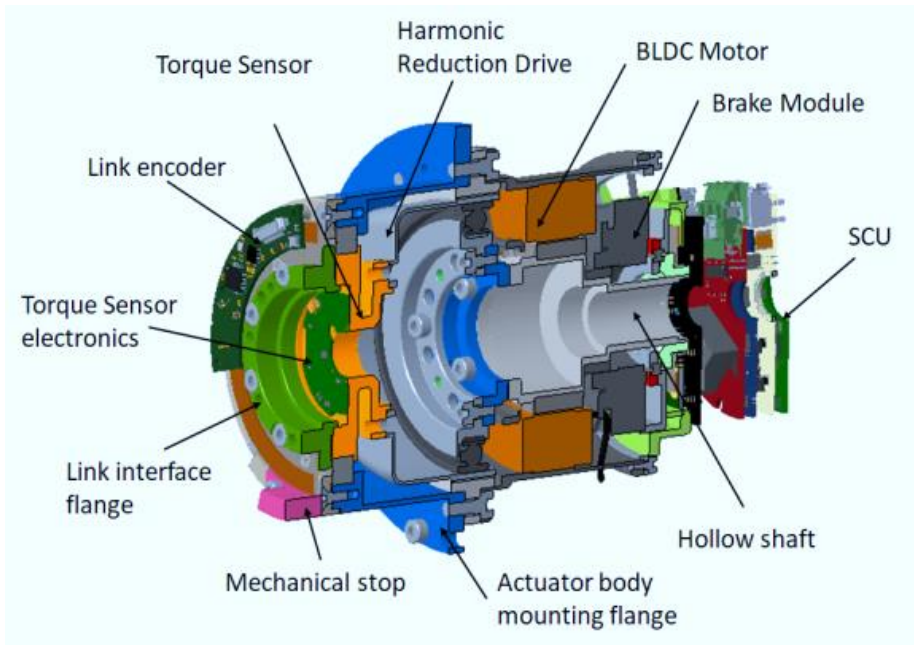




# Breadboard System Design

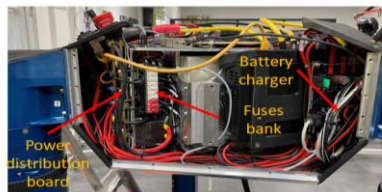
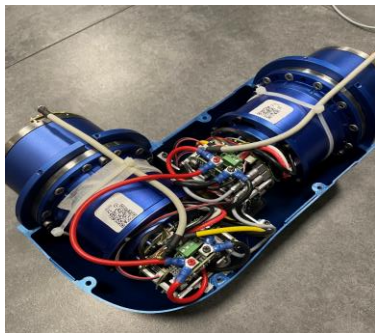
## Relocatable Manipulator

- Manipulator based on two actuator set sizes →



# Breadboard System Design

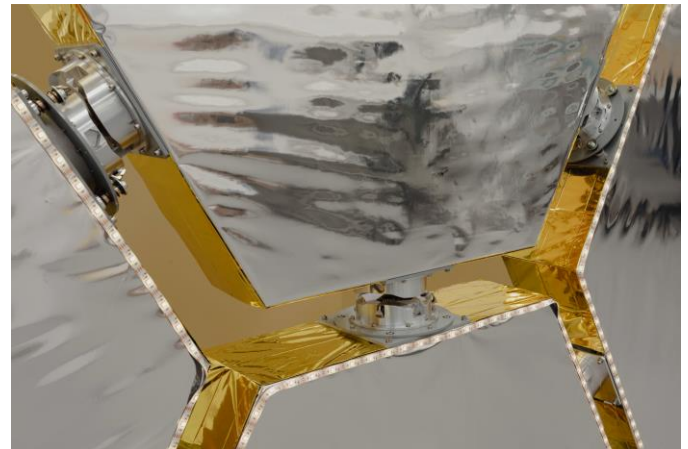
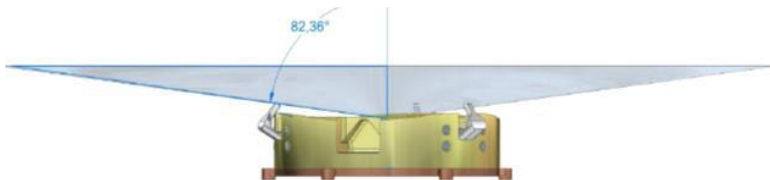
## Relocatable Manipulator



# Breadboard System Design

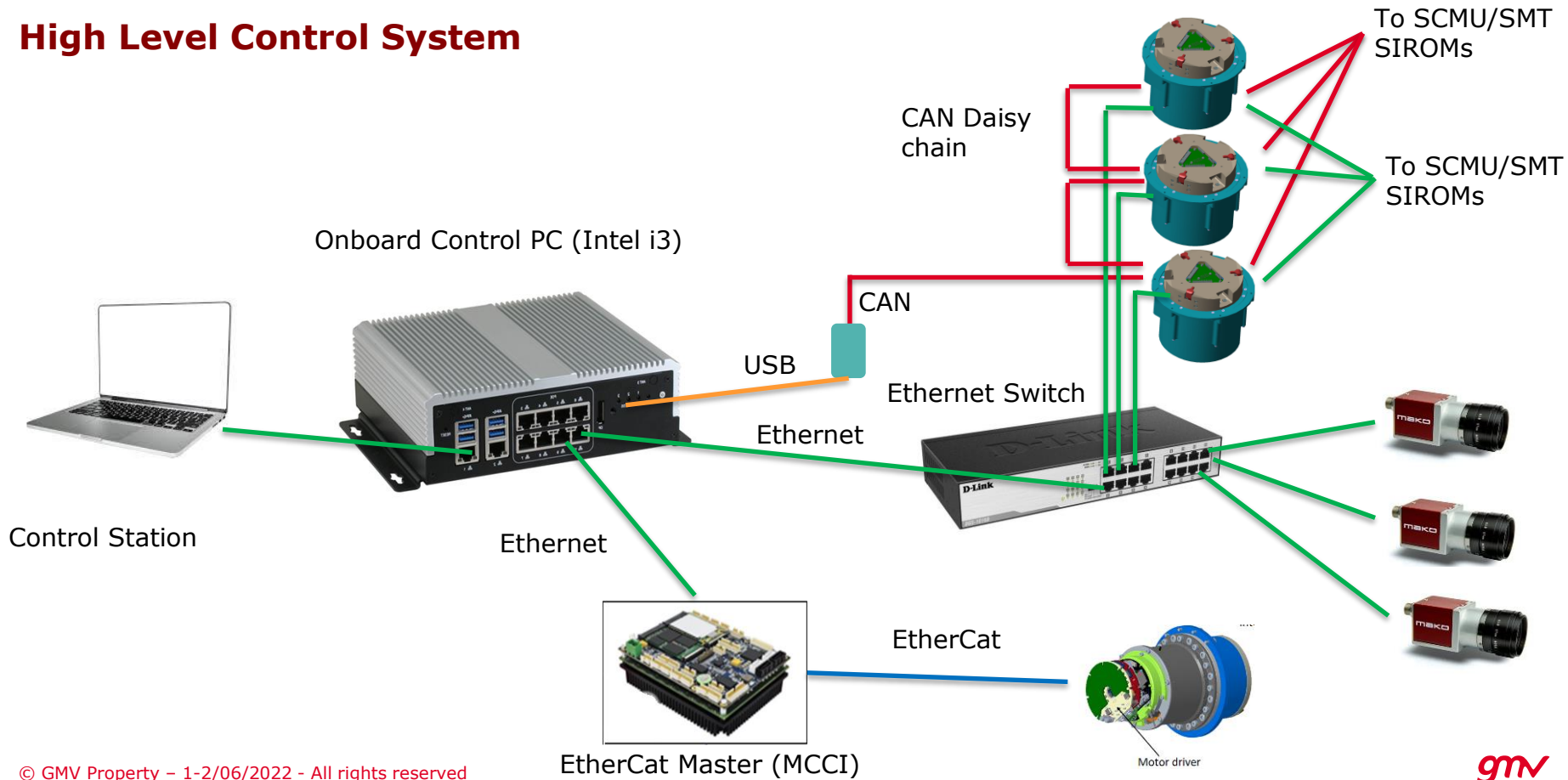
## Standard Interconnect (SIROM)

- SIROM (Standard Interface for Robotic Manipulation)
- Integrated interface: mechanical, data, electrical and fluid connectivity
- Evolution of a TRL-6 H2020 development:
  - Guiding surfaces allowing docking with angle  $>60^\circ$
  - Removal of fluid interface
  - Higher deliverable latches force
  - Lines for high power transmission
  - Active and Passive versions (1,1kg and 0,43 Kg)



# Breadboard System Design

## High Level Control System





# Breadboard System Design

## Vision System

3 Cameras and lights near end-effectors

3D fiducial markers

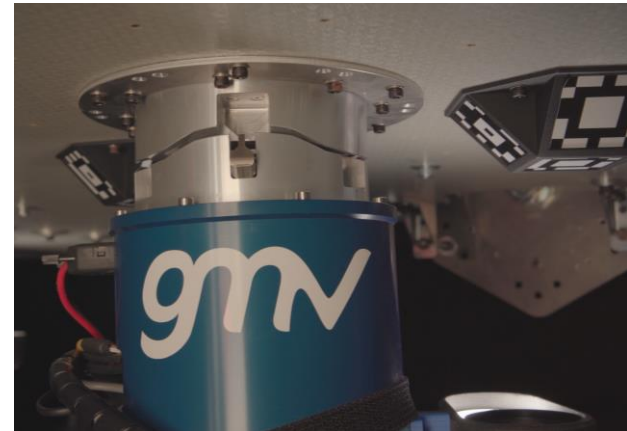
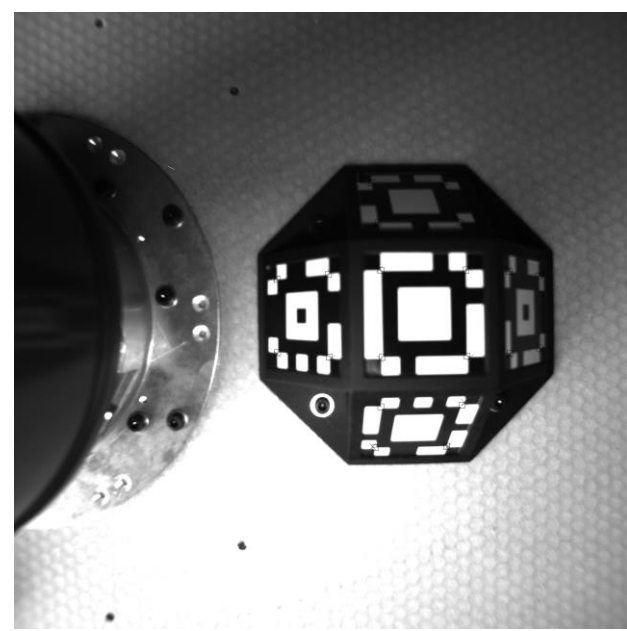
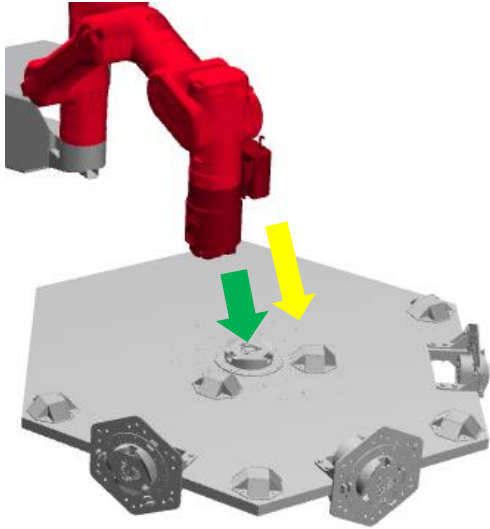
Image Based Visual Servoing:

- Eye-in-hand Visual Servoing: Manipulator Attach
- Eye-in-head (external): SMT Assembly



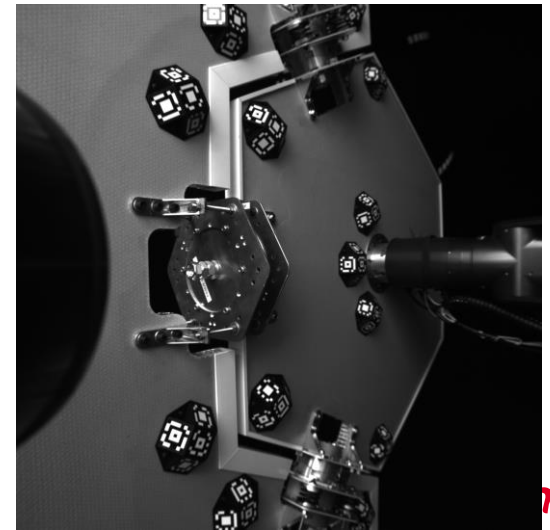
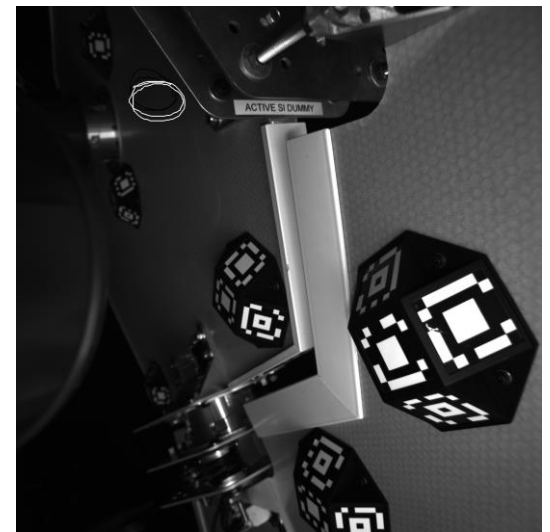
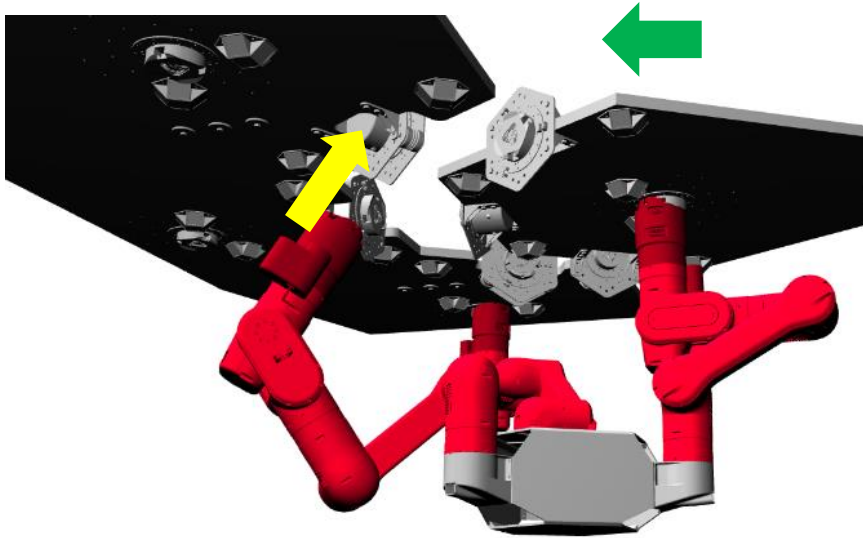
# Breadboard System Design

## Visual Control: Eye-in-hand Visual Servoing



# Breadboard System Design

## Visual Control: External visual servoing



# Breadboard System Design

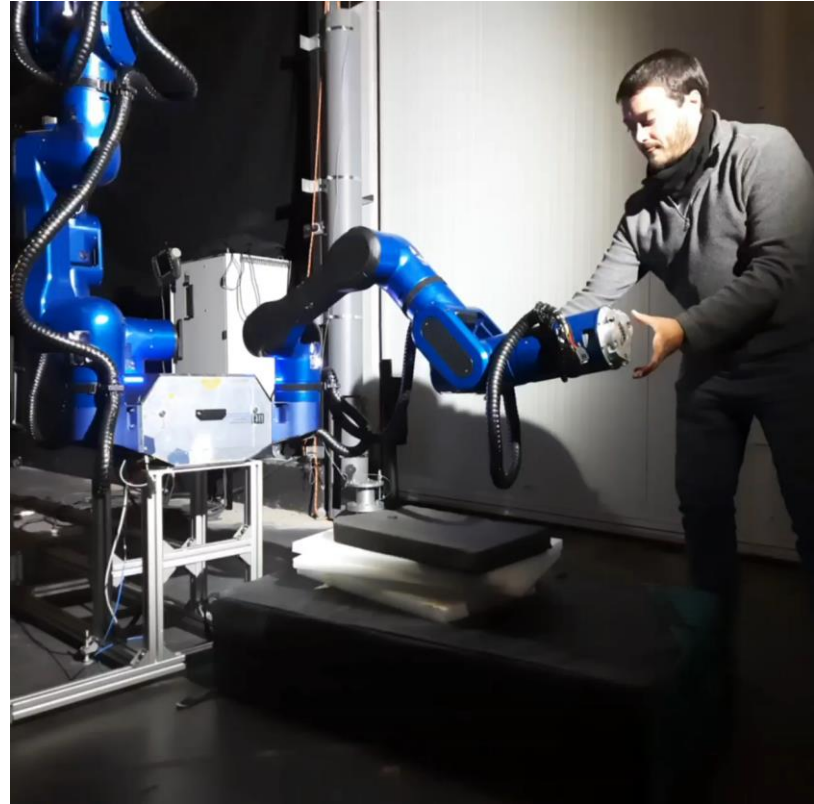
## Force-based Control

### Impedance Control (IC)

Gravity compensation used for ground testing

### Two IC use cases:

- Safe Visual approach for SMT Assembly and Manipulator Attach
- Force relaxation for SIROM latching





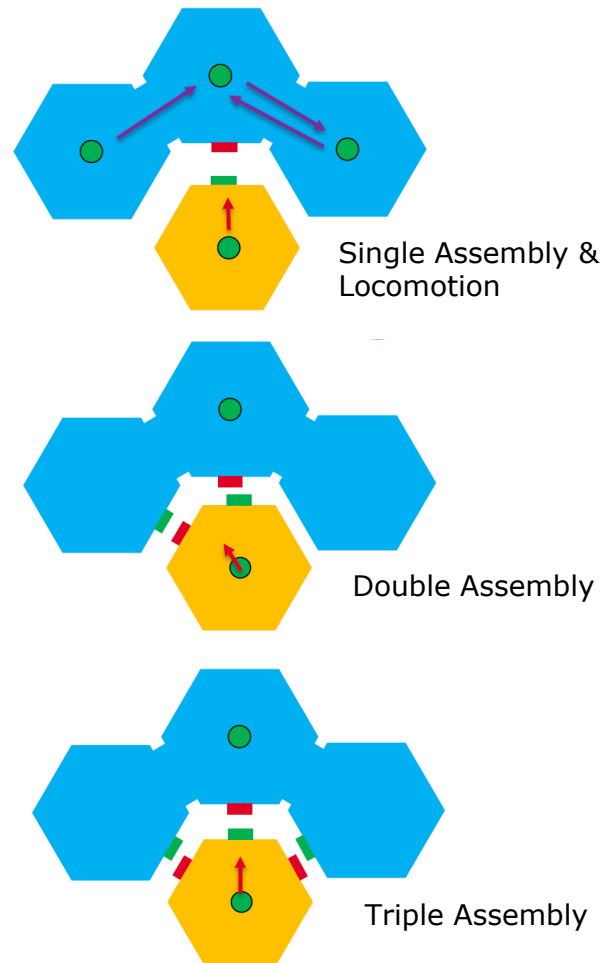
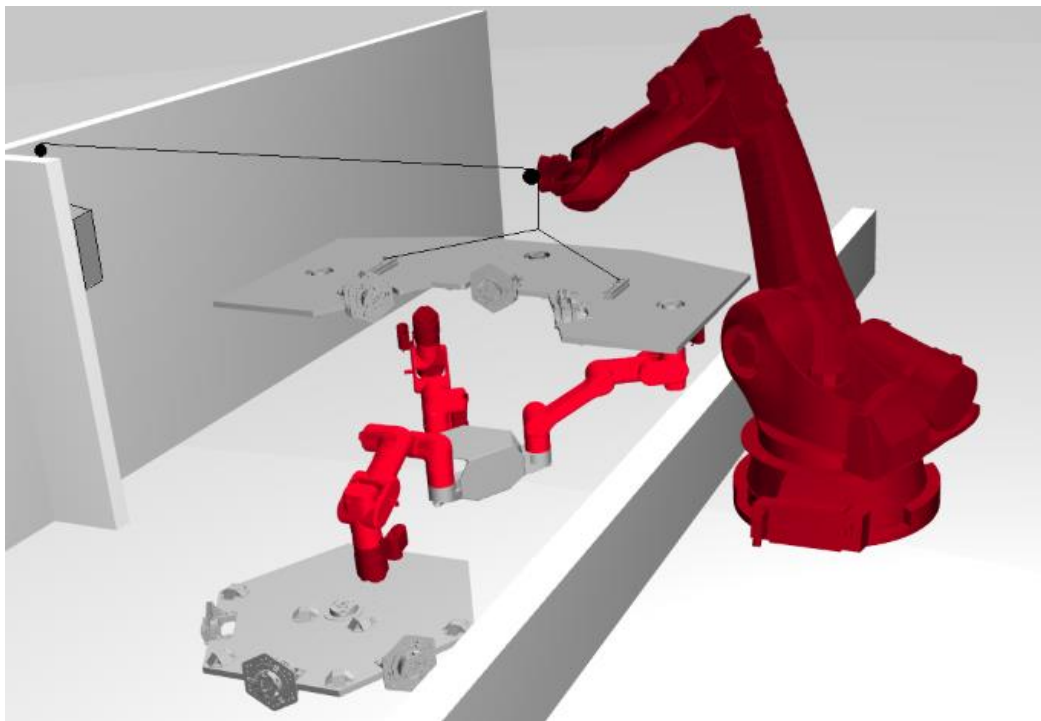
# Contents

- Mission and Concept of Operation
- Breadboard System Design
- Test Plan
- Testbed Design
- Test Results
- Conclusions



# Test Plan

## Ground Test Setup and main operations to be tested



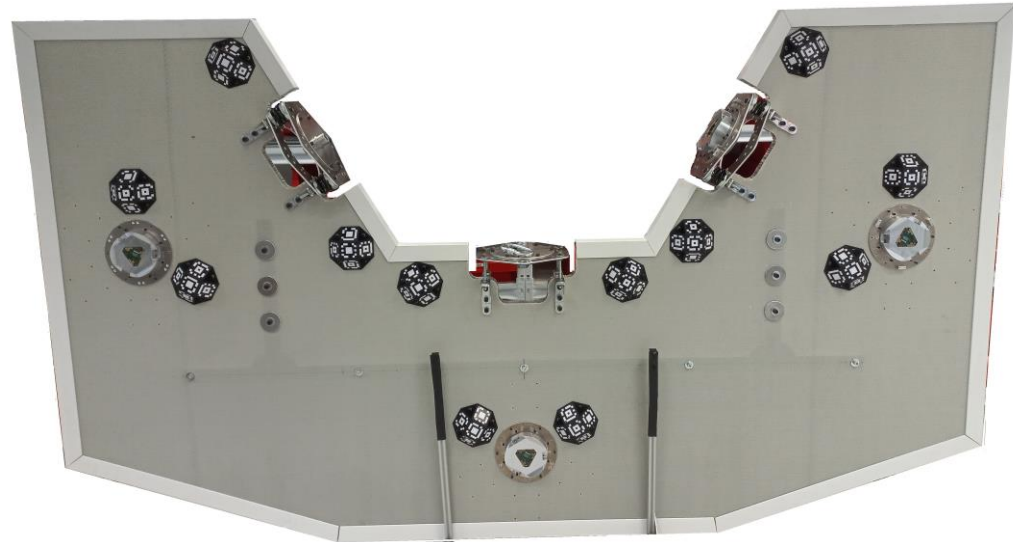
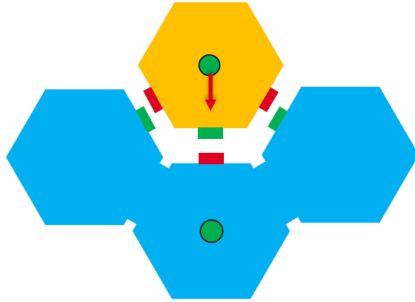
# Testbed Design

## Telescope/spacecraft Mock-ups

### Single Mirror Tile and Telescope mock-ups

(12Kg and 25 Kg) including:

- Lightweight honeycomb panels structure
- Assembly SIROMS
- Manipulation/Locomotion SIROMS
- Fiducial markers

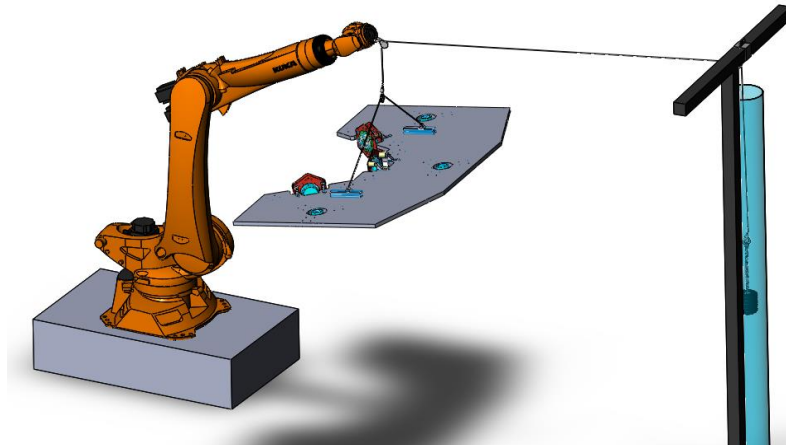
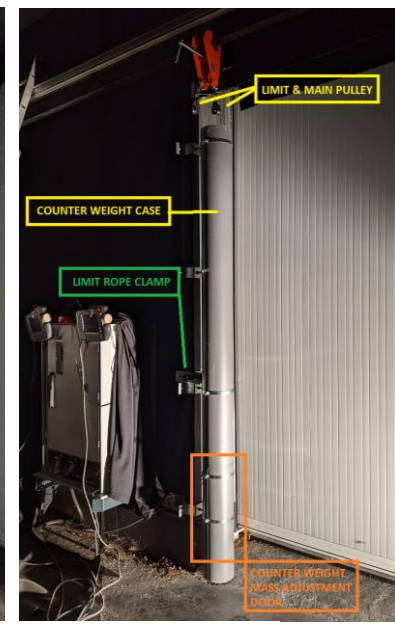


# Testbed Design

## Weight Compensation Device

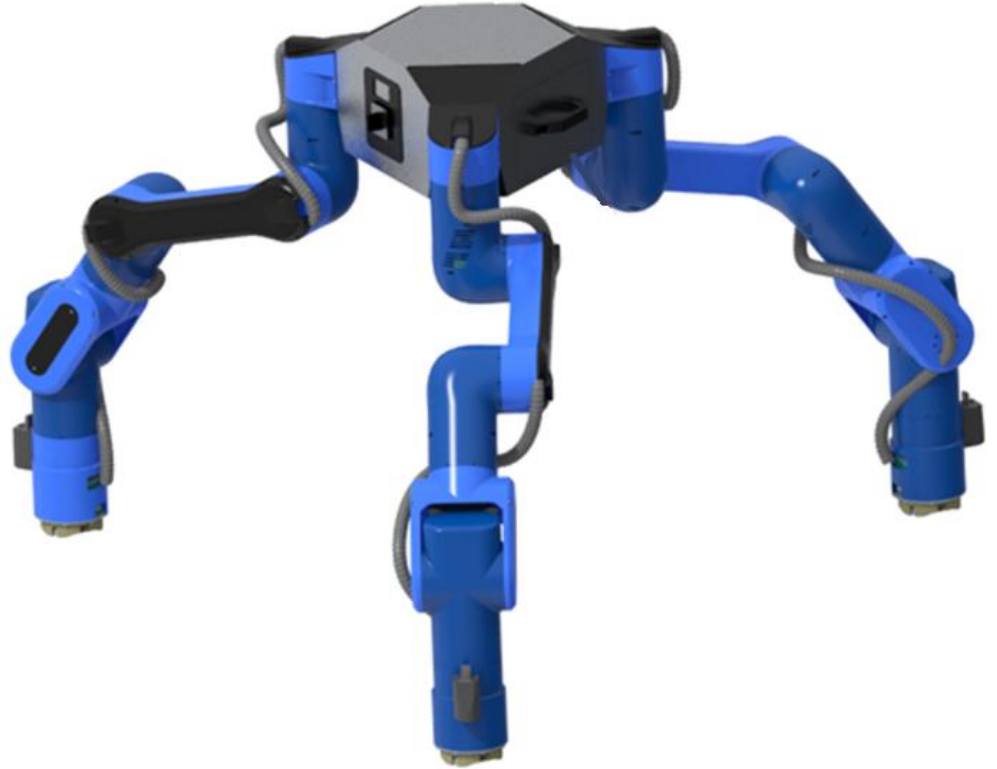
### Weight Compensation Device:

- Offloads the Telescope mock-up (25 Kg)
- WCD manipulator tracks motion of CoG of telescope mock-up
- Steel cable, pulleys and counterweight compensate weight
- Cable setup allows rotations around the telescope CoG



# Contents

- Mission and Concept of Operation
- Breadboard System Design
- Test Plan
- Testbed Design
- Test Results
- Conclusions

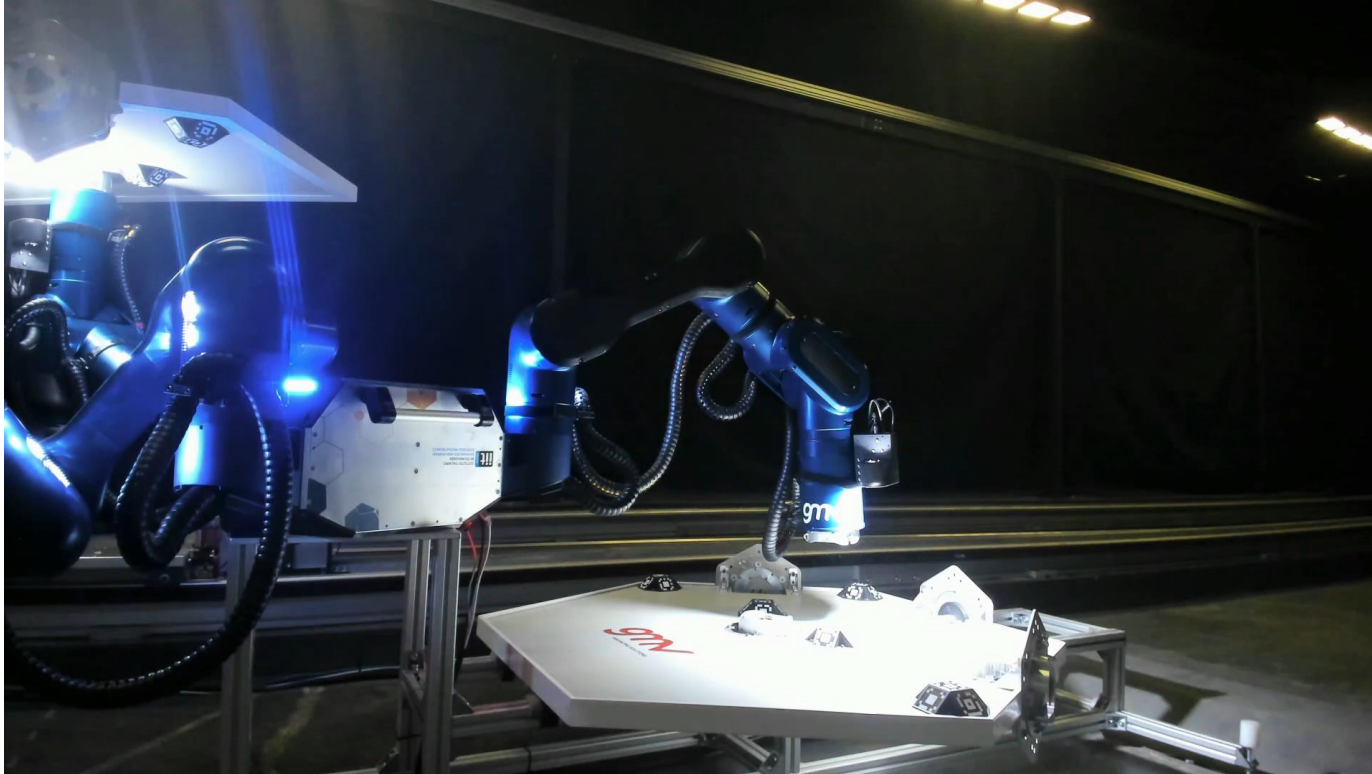


# System Testing

## Manipulation Test

### Grasp operation phases:

1. Free motion approach
2. Visual servoing approach
3. IC Compliant latching





# System Testing

## Locomotion Test (I)

### Locomotion operation phases:

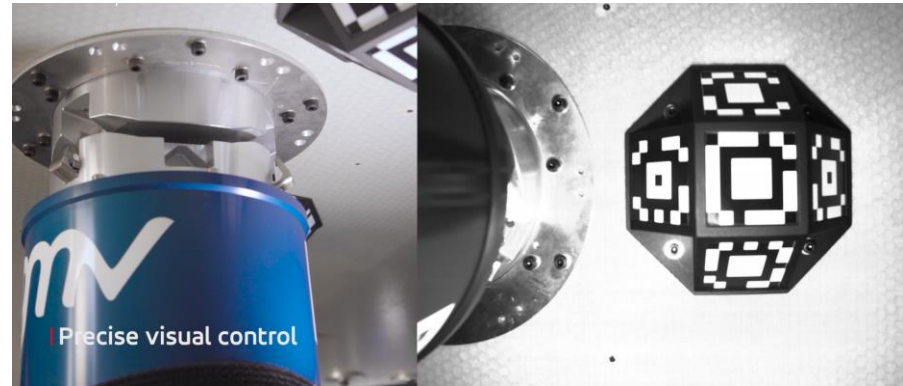
1. Free motion approach
2. Visual servoing approach
3. IC Compliant latching
4. Unlatching (another arm)
5. Body motion



# System Testing

## Locomotion Test (II)

- **Test:** Robot waks on the three manipulation SIROMS of the telescope
  - o Weight Compensation Device used
- **Results:** all tests successful:
  - o Very reliable operation (over 40 steps performed)
  - o Accurate and robust eye-in-hand visual servoing (5 mm accuracy needed)
  - o Smooth operation of the WCD





# System Testing

## Assembly Test (I)

### Assembly operation phases:

1. Free motion approach
2. Visual servoing camera approach → eye-in-hand VS
3. Visual servoing assembly approach → external VS
4. IC for compliant latching



# System Testing

## Assembly Test (II)

- **Tests:** Single / Double / Triple Assembly and Dissassembly
- **Results:** All tests successful:
  - The position/orientation accuracy needed (3mm / 1 deg) was reached consistently
  - Collisions can be expected, but they are not problematic  
→ Use of IC + VS is recommended
  - Visual servoing is stable, robust and accurate, even in presence of structure vibrations
  - Consistent, repeatable results, obtained in 5 test repetitions

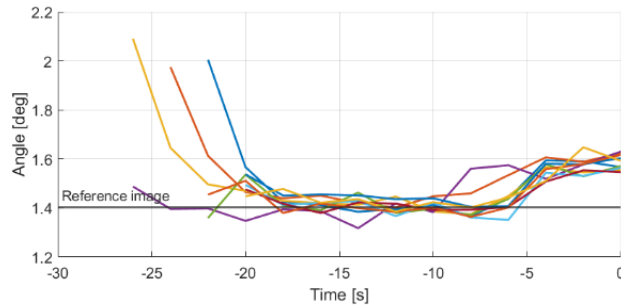
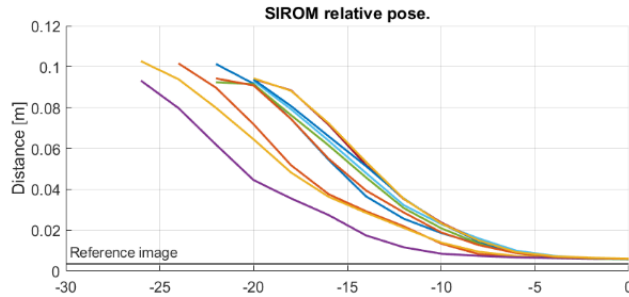


# System Testing

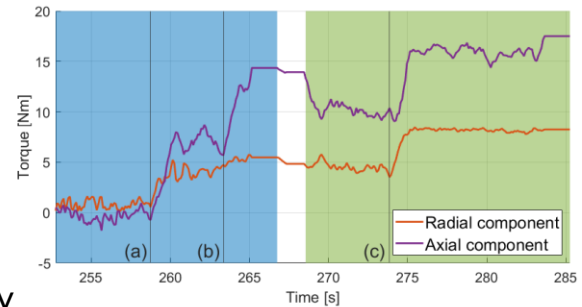
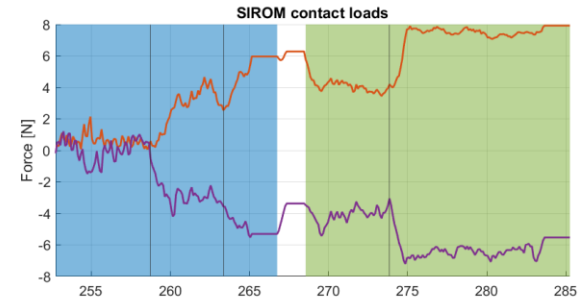
## Assembly Test (III)

### Assembly operation phases:

1. Free motion approach
2. Visual servoing camera approach → eye-in-hand VS
3. Visual servoing assembly approach → external VS
4. Compliant latching



Final error: 2.7 mm, 0.17 deg



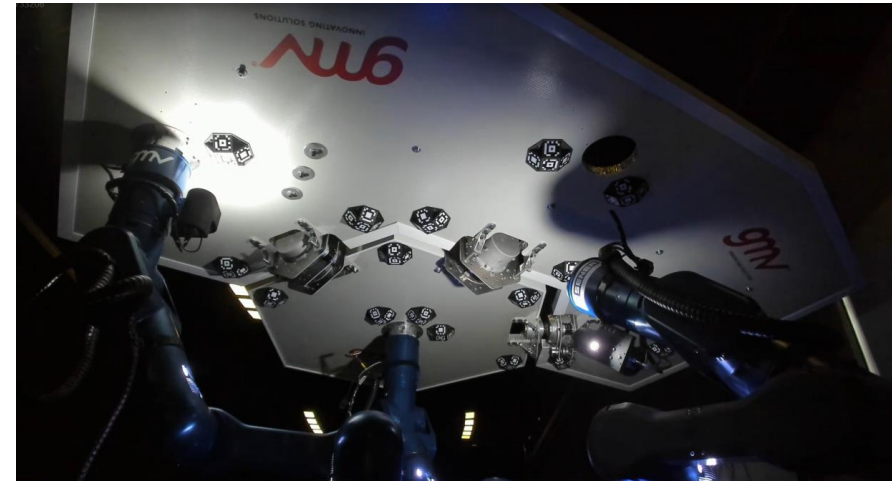
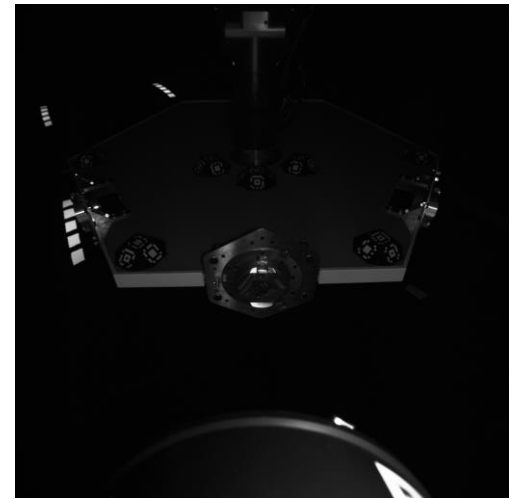
Assembly phase:

1./2. 3. 3. 4

# System Testing

## Light Conditions Robustness Test

- **Tests:** Triple assembly with lighting conditions:
  - o Diffuse light
  - o Darkness
  - o Focused light near telescope plane
- **Results:** All tests successful:
  - o Best results in darkness (eclipse)
  - o Lighting parallel to telescope is the worst case
  - o Changing light conditions can result in unstable motion
  - o Exposure time adjusted manually
    - Auto-exposure + HDR needed in a flight system
    - A "smart" autoexposure optimising visibility of markers is proposed for a flight system

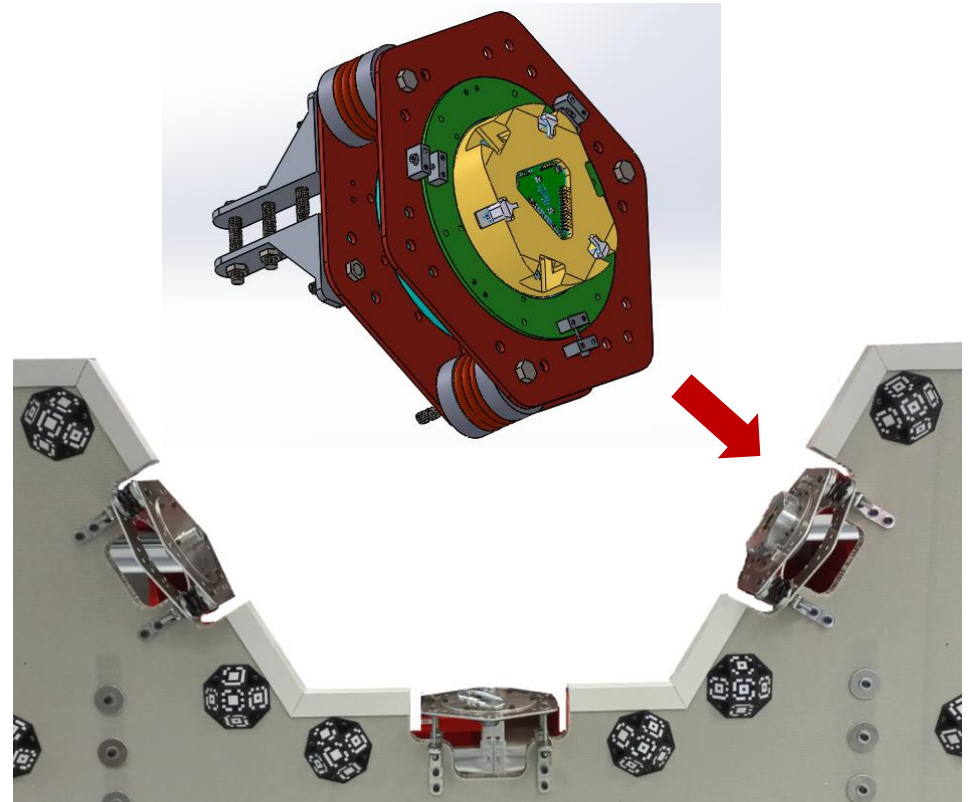


# System Testing

## Structure Misalignment Robustness Test



- **Tests:** Triple assembly with SIROM misalignment conditions:
  - o 0 mm misalignment
  - o 1 mm misalignment
  - o 2 mm misalignment
- **Results:** Successful for misalignment < 2mm
  - o Failure to complete the latching (mechanically) for 2 mm misalignment
  - o MIRROR robot performance is good even in misalignment conditions



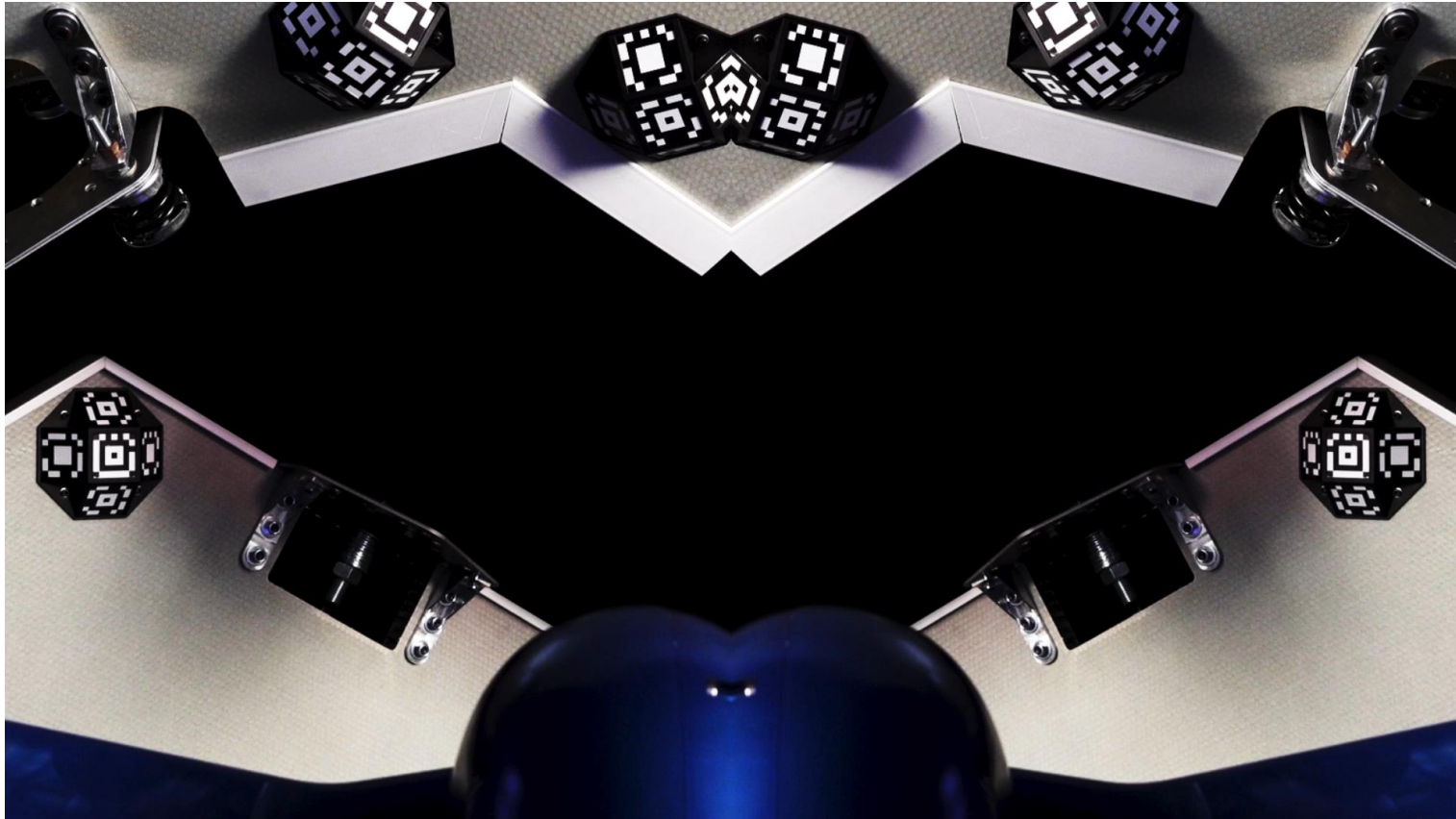
# Conclusions



## MIRROR activity conclusions and way forward

- The whole concept for a re-locatable manipulator for on-orbit assembly has been validated successfully, including:
  - o The sensor approach, based on three cameras and joint torque sensors
  - o The vision-based and force-based control approach for precise and safe manipulation and assembly
- The breadboard system has been developed and tested in ground up to TRL4
- Next steps should address aspects not covered in the activity:
  - o On-orbit representative testing conditions for vision, Autoexposure+HDR, etc.





# Thank you

MIRROR Team

