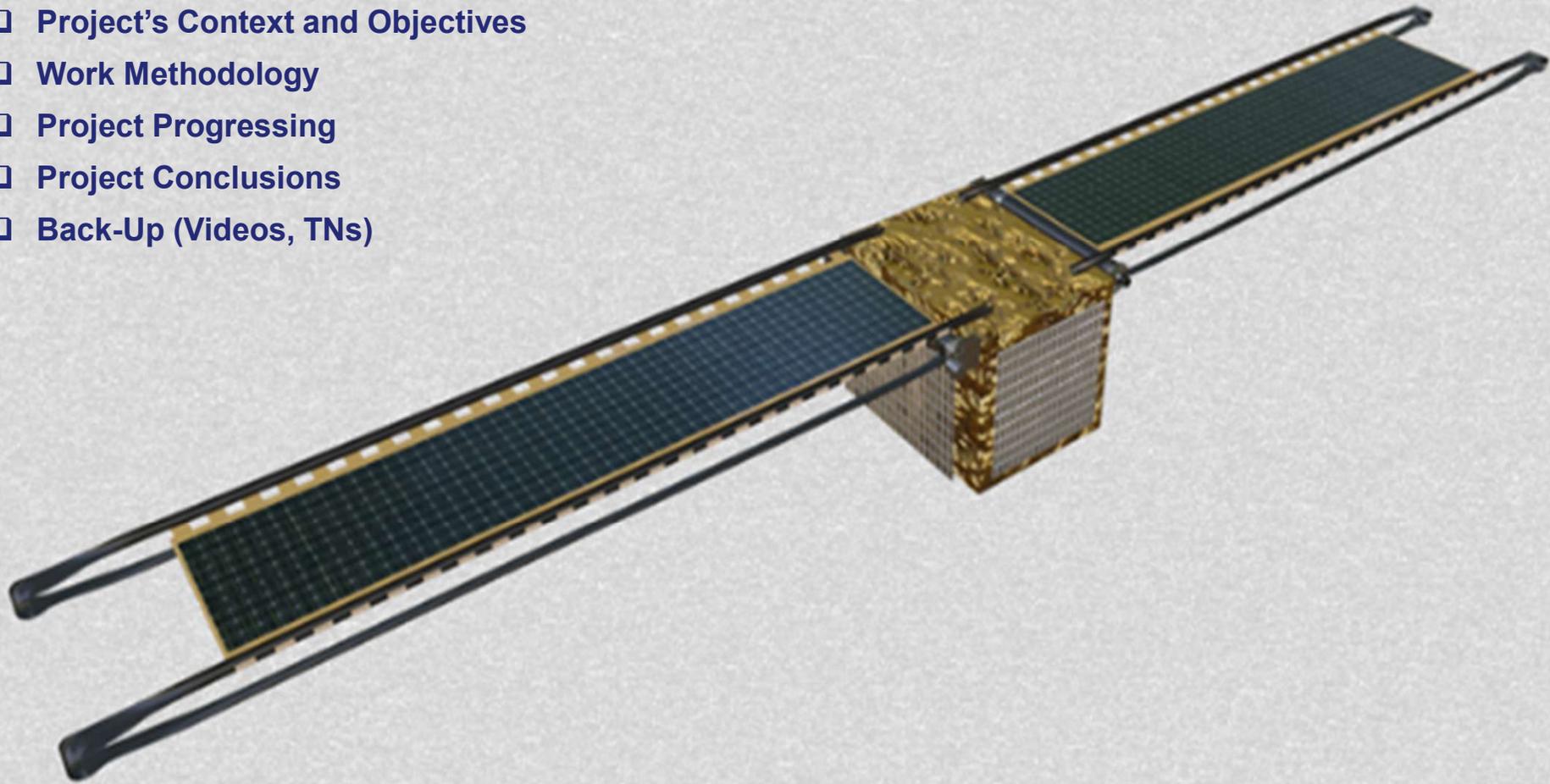


DEVELOPMENT OF NEW TYPE OF FLEXIBLE SOLAR ARRAYS FOR CONSTELLATIONS UP TO 5KW

FINAL REPORT PRESENTATION, 16/12/2024

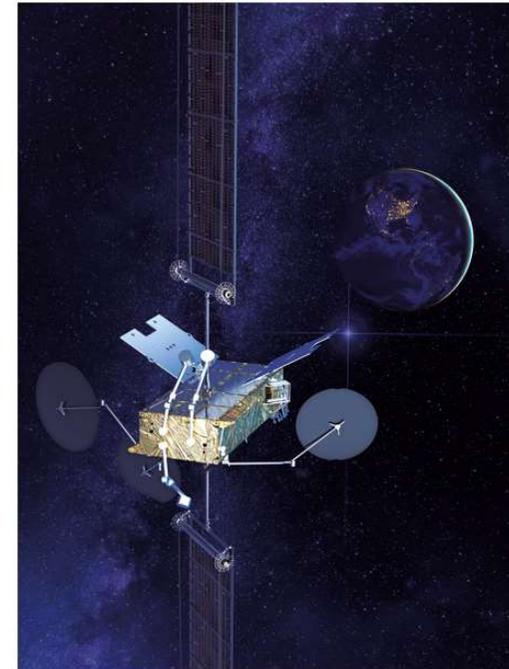
- Project's Context and Objectives
- Work Methodology
- Project Progressing
- Project Conclusions
- Back-Up (Videos, TNs)



PROJECT'S CONTEXT AND OBJECTIVES

- ❑ In recent years, much progress has been made in the development of high performance solar arrays, primarily for GEO applications.
- ❑ This study has been performed in order to assess the suitability of the SolarFlex concept for adaptation to LEO constellation applications:
 - ❑ with power requirements up to 5 kW.
 - ❑ with low cost
 - ❑ with the possibility to be packed tightly into limited stowed volumes (target range 25-40 kW/m³).

SolarFlex Product of Thales Alenia Space

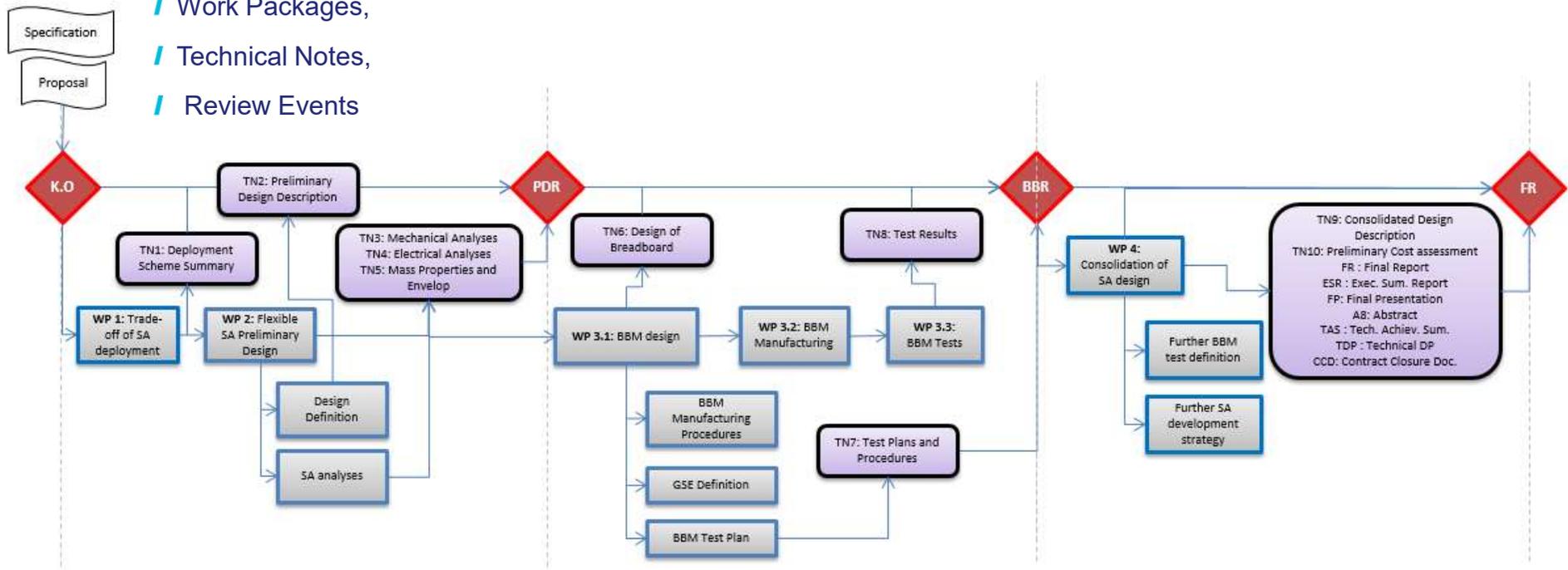


Fully flexible SA of Thales comprises blankets rolled around a mandrel, onto which the photovoltaic assembly is bonded, deployed by a pair of tape springs on either side

WORK METHODOLOGY

/// Work Logic:

- // Work Packages,
- // Technical Notes,
- // Review Events



WORK METHODOLOGY - DOCUMENTS DELIVERY/ACCEPTANCE STATUS

Document Identifier and title	Document Reference	Issue	Delivery/acceptance status
TN1- Deployment Scheme Summary	0005-0015797095	01	Delivered and accepted
TN2 - Preliminary Design Description	0005-0017470580	01	Delivered and accepted
TN3 – Mechanical Analyses	0005-0016711540	02	Delivered and accepted
TN4 – Electrical Analyses	0005-0016616955	02	Delivered and accepted
TN5- Mass Properties and Envelop	0005-0016699396	01	Delivered and accepted
TN6 – Design of Breadboard	0005-0017754749	01	Delivered and accepted
TN7 – Test Plans and Procedures	0005-0018693636	01	Delivered and accepted
TN8 – Test Results	0005-0018693664	01	Delivered and accepted
TN9 - Consolidated Design Description	0005-0018697286	01	Delivered and accepted
TN10 – Preliminary Cost Estimation	0005-0018896086	01	To be delivered by mail today 16th December
TDP - Technical Data Package	-	01	To be delivered by mail today 16th December
AB - Abstract	0005-0018906479	01	To be delivered by mail today 16th December
TAS - Technology Achievement Summary	0005-0018905980	01	To be delivered by mail today 16th December
FP – Final Presentation	0005-0018899372	01	To be delivered by mail today 16th December
ESR - Executive Summary Report	0005-0018905811	01	To be delivered by mail today 16th December
FR - Final Report	0005-0018899165	01	To be delivered by mail today 16th December
CCD - Contract Closure Documentation	-	-	To be delivered by mail today 16th December

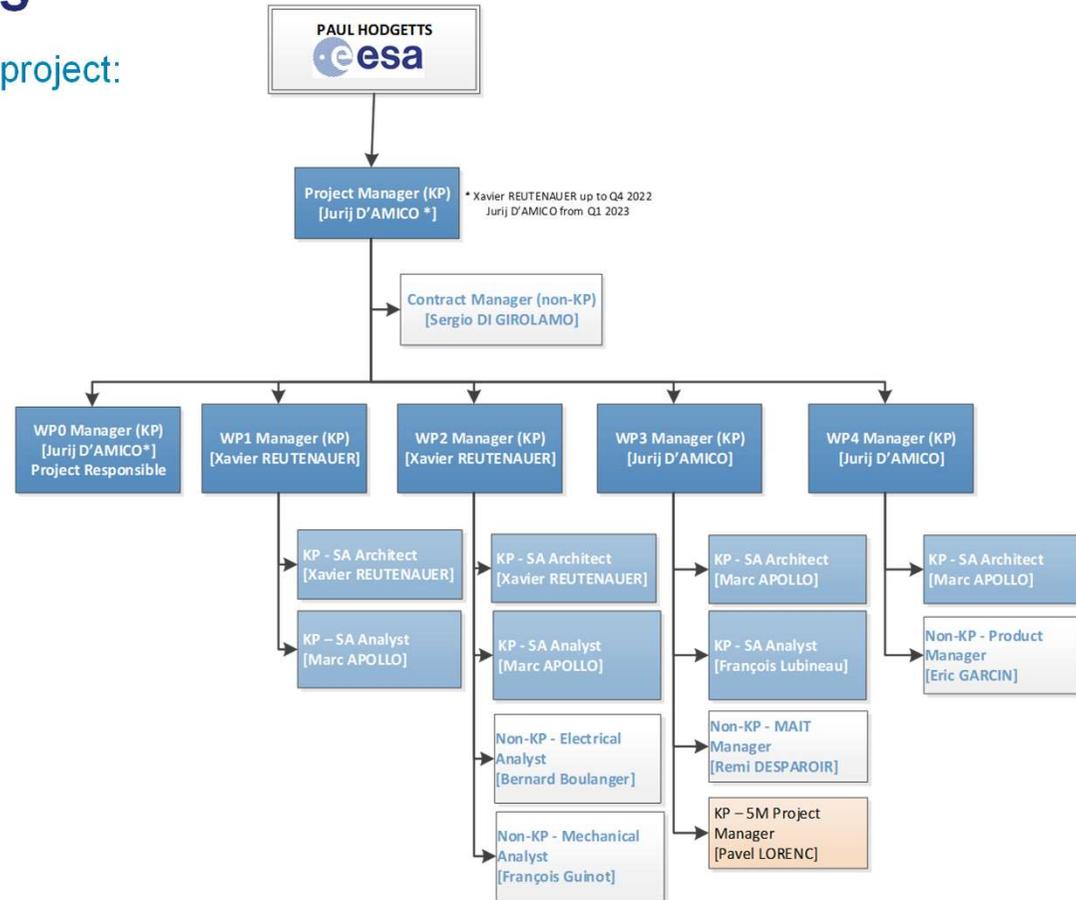
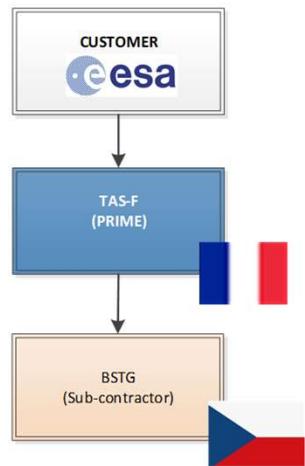
WORK METHODOLOGY – WORK PACKAGE AND TASKS

/// The 4 Work Packages (“WPs”) includes the following tasks:

- ! WP1: *Flexible SA Concept Trade-off*, focused on the investigation of different deployment schemes.
- ! WP2: *Flexible SA design and analysis*, focused on the:
 - preliminary design of solar array concept.
 - prediction of electrical performance.
 - mechanical analyses.
 - estimation of mass and inertia.
- ! WP3: *Flexible SA Breadboard Design & MAIT*, focused on the design, manufacturing and testing of breadboard models.
- ! WP4: *Flexible SA consolidated design & perspectives*, focused on the conclusions and way-forwards

WORK METHODOLOGY - OBS

/// Participating companies involved in this project:



PROJECT PROGRESSING – REAL TIMELINE

/// The duration of the work is 34 months from kick-off

TDE 5kW Activity	Year	2022												2023												2024						2025													
	Month	1	2	3	4	5	6	7	8	9	10	11	12	1	2	3	4	5	6	7	8	9	10	11	12	1	2	3	4	5	6	7	8	9	10	11	12	1	2	3	4	5	6		
	Lead time			1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25	26	27	28	29	30	31	32	33	34	35	36	37	38	39	40		
KO				▼																																									
WP1: Investigation Phase																																													
WP2: Design Phase																																													
WP3.1: Manufacture Phase																																													
WP3.2: Test Phase																																													
WP4: Consolidation Phase																																													
Deployment Scheme Review																																													
PDR (Part 1)																																													
PDR (Part 2)																																													
PM 1																																													
PM 2																																													
PM 3																																													
PM 4																																													
TRR																																													
TRB																																													
PM 5																																													
FR																																													
FP																																													

PROJECT PROGRESSING – WP1 MAIN ACHIEVEMENTS

/// This phase has included the following achievements:

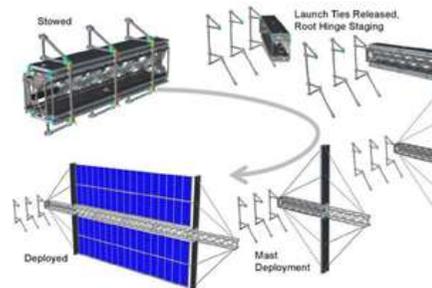
- / Investigation of the features of the following 4 deployment schemes:
 - Fan folded solar array
 - Accordion SA with central mast deployment on panel rear face
 - Accordion SA with central mast deployment
 - Roll out SA with tape spring for deployment
- / Selection of the “Roll out solution” as the most promising architecture for adaptation in terms of cost, low stowage volume, simplicity of release/activation while meeting the power demands of up to 5kW



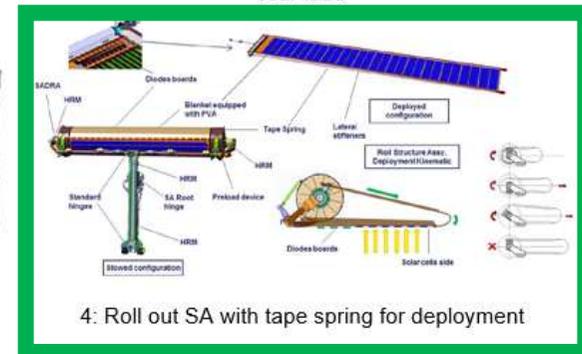
1: Fan folded solar array



2: Accordion SA with central mast deployment on panel rear face



3: Accordion SA with central mast deployment on panel rear face



4: Roll out SA with tape spring for deployment

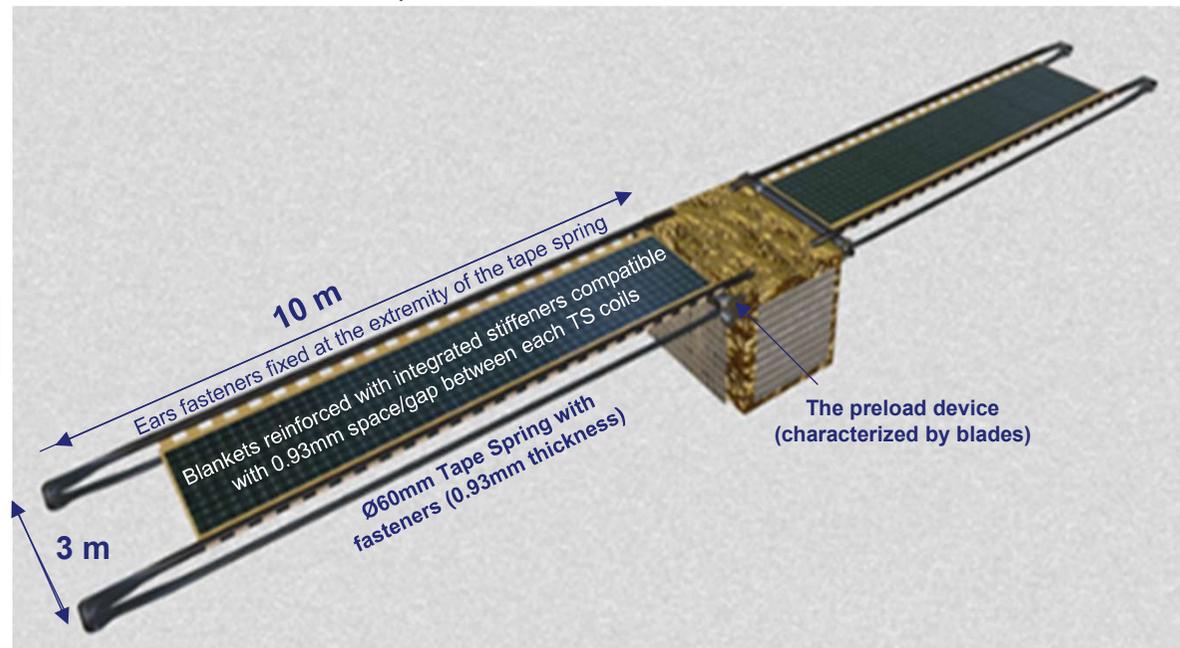
SELECTED

PROJECT PROGRESSING – WP2 MAIN ACHIEVEMENTS

/// Based on a preliminary array design and representative breadboard models, the Thales team has demonstrated the feasibility of the development of a fully flexible variant of SolarFlex product:

1 wing	Silicon cells	AsGa cells
Power EOL [kW]	2.85	4.9
Volume [kW/m ³]	33	57
Power/Mass [W/Kg]	57	98

Three-dimensional concept of the miniaturized flexible SA for LEO constellations



PROJECT PROGRESSING – WP3 MAIN ACHIEVEMENTS

/// The miniaturized architecture has been manufactured and tested in order to verify its deployment capability.

The act of motorizing and bringing GS BB from stowed to deployed configuration is successfully performed

/// An high confidence on the suitability of the proposed innovative technologies for deployment has been built. It is the case of preload devices characterized by blades, synchronization and deployment devices based on fasteners and pins.

/// However to overcome visible anomalies on the blankets (folds during deployments and through the bulb passage) design optimization and additional tests are recommended, in order to improve the tensioning and flatness of the blankets and co-alignments of the tapes springs.

SA BB deployment test (integrated on 0G trolley placed in the clean room under rails)



PROJECT PROGRESSING – WP4 MAIN ACHIEVEMENTS

- /// The final task has allowed to update the risk register for next activities.
- /// A preliminary cost estimation of the risk mitigation plan is defined for rapid implementation (650,000 EUR is the estimated cost to achieve the TRL5)

Risk	Point	Domain	Problem Description	Mitigation strategy
Risk 1 / Risk 12	Point 1	Flexible GS BBM Integration/Assembly	Difficulties were encountered during BBM assembly process in stowed configuration (misalignments and synchronization problem between BBM components along the thickness and through the deployment axis)	Integration/Assembly in deployed configuration seems mandatory, including rigid stiffeners in the assembly process
	Point 2	Flexible GS BBM Synchronization System Functioning	Operation problem was encountered during the assessment of the synchronization systems functioning (for back side) (misalignments between TS holes and pin thickness prediction locations due to TS manufacturing defects)	Control of the TS drilling plan based on better TS thickness prediction
Risk 1	Points 3 and 4	Flexible GS BBM Fasteners developments	Space material for fasteners is not available. Consequently a development is needed. Sizing of fasteners is simply based on preliminary experimental de-risking test done at TS level. Incompatibility with stiffeners positioning.	Consolidation of the fasteners development strategy based on suitability with numbers and positions of external rigid stiffeners (see problems 4 and 5).
	Point 5	Flexible GS BBM deployment positioning during deployment)	Deployment problem of the TS observed from the unsuitable positioning of the GS (not controlled spacing between TS)	Replacement of stiffeners integrated/embedded in the blanket with external rigid stiffeners including an optimal BBM integration plan.
	Point 6	Flexible GS BBM deployment positioning during deployment)	Deployment problem of the blanket observed through the bulb passage (Blanket folds appears)	
	Point 7	SA positioning during rolling	Rolling problem of the GS observed from the unsuitable synchronization of the TS and insufficient effort on preload and preload devices (with additional rollers). system	Improvement of synchronization system (with pins)
	Point 8	SA Global Architecture	Integrated Y stiffeners in blankets and blanket ears are not suitable to avoid problem during deployment (see points 5 and 6).	The integration of external rigid stiffeners is essential as long as the critical evaluation of their compatibility with fasteners and other possible impact on the design for their fixation



/// 12 Example of risk management tool to identify potential setbacks within the project

Date: 16/12/2024
 Ref: 0005-0018899372
 Template: 83230347-DOC-TAS-EN-009

This document is not to be reproduced, modified, adapted, published, translated in any material form in whole or in part nor disclosed to any third party without the prior written permission of Thales Alenia Space.
 © Thales Alenia Space, 2020 All right reserved

THALES ALENIA SPACE INTERNAL

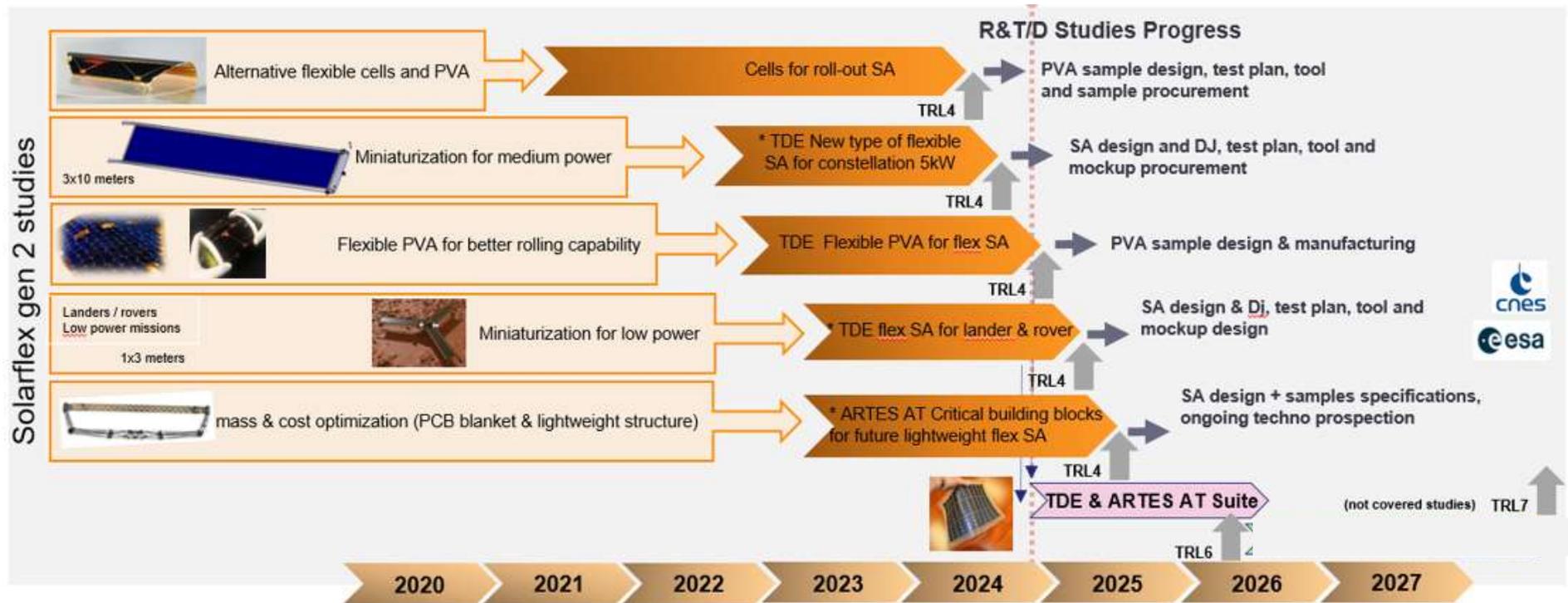


CONCLUSIONS

/// The suitability of the SolarFlex concept for adaptation to LEO constellation applications can be confirmed based on the following advantages:

- / An extremely compact design (possibility to be packed tightly into stowed volumes range 33-57 kW/m3).
- / Compliant power capacity per wing (range 2.85 - 4.9 kW)
- / A proven deployment functioning
- / A budget less than 1 MEUR is envisaged to increase the TRL up to 5.

BACK-UP - STRATEGY



THANKS !

QUESTIONS?

